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# PREPWATE—An Interactive Preprocessing Computer Code to the Weight Analysis of Turbine Engines (WATE) Computer Code

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UTTL: PREWATE: An interactive preprocessing computer code to the Weight  
Analysis of Turbine Engines (WATE) computer code

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MAJS: /\*COMPUTER PROGRAMS/\*PREDICTIONS/\*TURBINE ENGINES/\*TURBOFAN ENGINES/\*  
WEIGHT (MASS)

MINS: / COMPRESSORS/ DESIGN ANALYSIS/ THERMODYNAMIC PROPERTIES/ THRUST

ABA: Author

ABS: The Weight Analysis of Turbine Engines (WATE) computer code was developed  
by Boeing under contract to NASA Lewis. It was designed to function as an  
adjunct to the Navy/NASA Engine Program (NNEP). NNEP calculates the design  
and off-design thrust and sfc performance of User defined engine cycles.  
The thermodynamic parameters throughout the engine as generated by NNEP  
are then combined with input parameters defining the component  
characteristics in WATE to calculate the bare engine weight of this User  
defined engine. Preprocessor programs for NNEP were previously developed  
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similar preprocessor for the WATE code.

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PREPWATE - AN INTERACTIVE PREPROCESSING  
COMPUTER CODE TO THE WEIGHT ANALYSIS OF TURBINE  
ENGINES (WATE) COMPUTER CODE

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SUMMARY

The Weight Analysis of Turbine Engines (WATE) computer code was developed by Boeing under contract to NASA Lewis (ref. 1). It was designed to function as an adjunct to the Navy/NASA Engine Program (NNEP) (ref. 2). NNEP calculates the design and off-design thrust and sfc performance of User defined engine cycles. The thermodynamic parameters throughout the engine as generated by NNEP are then combined with input parameters defining the component characteristics in WATE to calculate the bare engine weight of this User defined engine.

Preprocessor programs for NNEP were previously developed (ref. 3) to simplify the task of creating input datasets. This report describes a similar preprocessor for the WATE code.

INTRODUCTION

The NASA Lewis Research Center in conjunction with the Naval Air Development Center jointly developed a computer code (NNEP) for determining the thermodynamic performance for arbitrary turbine engines, i.e., the code can assemble arbitrary combinations of specified types of components (such as ducts, compressors, turbines, etc.) through the use of input variables rather than having to build a computer code exclusively for each engine configuration (ref. 2). In order to simplify the task of creating datasets to run on NNEP for new or occasional users, a pair of preprocessor programs KONFIG and REKONFIG were created (ref. 3). Favorable comments were received from users of the preprocessor programs. They also expressed the desire for a similar preprocessor for the Weight Analysis of Turbine Engines code (WATE) (ref. 1).

WATE was developed by Boeing under contract to NASA Lewis. It is capable of calculating the component weights and flowpaths of most engines that a user of NNEP would wish to study. It was designed to be an adjunct to NNEP and receives thermodynamic data from NNEP. Mechanical design variables are then set for each of the engine components and the weight and dimensions then determined.

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As opposed to NNEP, which treats all of a particular type of component (e.g., compressors) thermodynamically the same, WATE distinguishes them by type (e.g., fan, low pressure, high pressure). Different modules then calculate the weight by sub-type. The WATE code requires two main arrays of data, IWMEC and DESVAL. IWMEC is a two dimension array where the mechanical design inputs such as component type, the existence of frames, stators, and other information are stored. DESVAL is also a two dimensional array containing the aerodynamic and design inputs such as Mach number in, Mach number out, blade aspect ratio, etc. Default values of DESVAL for each component sub-type are built into PREPWATE. When the user runs PREPWATE, he will be shown the defaulted values as well as the recommended range for each variable. He may easily change one or more of the DESVALs for each component in the engine.

The rest of this report will show the operation of PREPWATE by a sample case.

#### OPERATION OF PREPWATE

PREPWATE consists of the main and one subroutine. The main routine prompts the User for all of the inputs required and prepares the output data-set. The subroutine (SUBROUTINE REDOEM) is called in order to allow the User to change one or more of the DESVALs from the defaulted values.

All reads are from UNIT 30 and prompts for input on UNIT 35. The final card images for the inputs to the WATE code are written on UNIT 6 and a data-set should be datadeffed to this unit to save the images.

#### SAMPLE CASE

A high bypass ratio separate flow engine will be used to illustrate the use of the code. The fan will have booster stages attached to it to raise the pressure of the air that will enter the core. These stages are driven by the low pressure turbine and the outer dimension of both turbines will be limited to the outer dimension of these booster stages. An NNEP thermodynamic schematic of the engine is shown in figure 1. As indicated, NNEP does not distinguish by subcomponent type, e.g., all compressor thermodynamic calculations are done identically.

The WATE code, on the other hand, requires that the User identify the type of subcomponent, e.g., the three compressors are now a fan, a low pressure compressor, and a high pressure compressor. The mechanical schematic for the WATE code for the NNEP engine in figure 1 is shown in figure 2. The component numbers are retained so there is a one to one correspondence with the thermodynamic model. Note, however, the inlet does not appear in figure 2. Inlet weight cannot be calculated by WATE. At NASA Lewis, this weight is calculated by the INSTAL computer code (ref. 4) which was also created as an adjunct to NNEP.

The User is now ready to run PREPWATE. A terminal session for creating the WATE input data for this engine is shown in Appendix A.

As can be seen from Appendix A, the user first defines a dataset on Unit 6 into which the card images for the WATE code will be written. He then identifies the library containing the Main Routine and Subroutine and then calls the main routine. (A procedure definition containing these instructions has been prestored in the LeRC 370/3033 as PREPWATE).

The User is then prompted for the required inputs. The first inputs specify what plots and whether English or SI units are desired. The second inputs are whether or not scaled engines are desired (IWT = 4) or not (IWT = 2) and for the amount of output desired (IOUTCD = 4 gives total weight, length and maximum radius, plus component values, plus stage by stage values).

Next, the disk weight calculation method to be used is identified (B = Boeing, G = Garrett). The User is then asked to identify the component numbers of all the components contributing to the maximum length of the engine (exclusive of inlets and water injectors). All the remaining flow components in the engine including the shafts, but excluding inlets and water injectors are then identified.

The User will then be prompted to enter the NNEP generic type of each component (COMP, TURB, NOZZ, etc.) and then for detailed information about each component (type of compressor, are there stators?, frames, etc.). After this information has been entered, the code will then print out the defaulted values for each DESVAL for each subcomponent as well as the recommended range (where appropriate). The User can change any of the defaulted values of DESVAL by entering the letter code for the variable followed by the new value which MUST include a decimal point. Any or all of the variables may be changed in this manner. By entering an "R", the User may review the latest values. When satisfied, the User enters a "T" to terminate this component and proceed to the next. When all the components are done, the program will terminate.

The card images generated from PREPWATE are shown in Appendix B. The extra blanks generated for DESVAL have been eliminated. This dataset must be incorporated into a NNEP dataset in order to run the NNEP/WATE combination. This dataset is also shown in Appendix B. A brief explanation of this dataset is as follows:

The first card image is the NNEP Title Card;

Next, the first block of Namelist D data. The defaulted values for all variables are being used.

The second block of Namelist D data for NNEP is then displayed. This represents the configuration and design spec data for Mode 1. (There is only one mode in this engine). There is an IWT = 4 card which tells the code that weight should be calculated after this NNEP point has been run. This triggers the code so that the WATE code is called.

The WATE code calls for the Namelist W data and the card images created by PREPWATE are entered here. The NNEP/WATE code is now ready to run. The answers to the sample case are shown in Appendix C. The interactive graphics capability of the code (ref. 5) was used to generate a graphics plot of the engine (PLOT = T in & W). This plot has been reproduced in figure 3. Note that the LPC (component 6) overlaps the duct (component 4). Changes to the DESVALs of components 2, 4 and/or 6 can be implemented to generate a better flowpath, either by going back to the original dataset or using the interactive method of reference 5 at the graphics scope.

The PREPWATE computer code listings are shown in Appendix D.

### CONCLUSIONS

PREPWATE has been created to help Users of the WATE code to prepare input datasets for the code. It is tutorial in nature and hopefully will prove to be as useful as the preprocessors KONFIG and REKONFIG for the NNEP computer code. The code is available from the author in card form ( 600 cards) or on a User supplied tape.



DDDEF FT06F001,VS, WATEDATA  
 DDDEF XX,VP,LIB.PREPDATE,OPTION C JOBLIB

GO:PREPDATE

ENTER VALUES FOR IPLT(PRINTER PLOT), PLOT(GRAPHICS PLOT), ISII(SI INPUT), ISIO(SI OUTPUT) -- T OR F

T T F F

ENTER VALUES FOR IWT (2=NO AIRFLOW SCALING,4=WITH AIRFLOW SCALING) AND  
 IOUTCD (0=WT,L, & MAX R,1=ADD COMPONENT WTS,2=ADD STAGE BY STAGE OUTPUT

4 2

DO YOU WANT BOEING OR GARRETT METHOD FOR DISK WEIGHT CALCULATIONS?  
 ENTER B FOR BOEING OR G FOR GARRETT

B  
 ENTER COMPONENT NUMBERS OF ALL COMPONENTS (EXCLUSIVE OF INLETS AND WINJS) THAT CONTRIBUTE TO MAX LENGTH OF ENGINE (RT. A  
 DJ.)

02 03 06 07 08 17 09 10 11 12 13 14  
 ENTER COMPONENT NUMBERS OF ALL REMAINING COMPONENTS (EXCLUSIVE OF INLETS,WINJS,LOADS,CHTLS,OPTVS, & LIMVS)

04 05 15 16  
 YOU WILL NOW BE PROMPTED FOR THE GENERIC (HNEP) TYPE FOR EACH COMPONENT (COMP,TURB, ETC.)  
 COMPONENT 2 IS A (A4)

COMP  
 COMPONENT 3 IS A (A4)

SPLY  
 COMPONENT 6 IS A (A4)

COMP  
 COMPONENT 7 IS A (A4)

DUCT  
 COMPONENT 8 IS A (A4)

COMP  
 COMPONENT 17 IS A (A4)

DUCT  
 COMPONENT 9 IS A (A4)

DUCT  
 COMPONENT 10 IS A (A4)

TURB  
 COMPONENT 11 IS A (A4)

DUCT  
 COMPONENT 12 IS A (A4)

TURB  
 COMPONENT 13 IS A (A4)

DUCT  
 COMPONENT 14 IS A (A4)

NOZZ  
 COMPONENT 4 IS A (A4)

APPENDIX A

DUCT  
COMPONENT 5 IS A (A4)

NOZZ  
COMPONENT 15 IS A (A4)

SHFT  
COMPONENT 16 IS A (A4)

SHFT  
FOR EACH NHEP COMPONENT YOU WILL BE PROMPTED FOR THE WATE SUBCOMPONENT  
COMPONENT 2 IS A COMPRESSOR, OPTIONS ARE (ENTER CORRECT LETTER)

- A- TYPICAL FAN
- B- OUTER PORTION OF NON-ROTATING SPLITTER FAN
- C- INNER PORTION OF NON-ROTATING SPLITTER FAN
- D- OUTER PORTION OF ROTATING SPLITTER FAN
- E- INNER PORTION OF ROTATING SPLITTER FAN
- F- LOW PRESSURE COMPRESSOR
- G- HIGH PRESSURE COMPRESSOR

A  
INDICATE WHETHER THERE ARE STATORS OR IF THIS IS A CENTRIFUGAL COMPRESSOR

- A- CALCULATE STATOR WEIGHT
- B- NO STATOR WEIGHT
- C- CENTRIFUGAL COMPRESSOR

B  
INDICATE WHETHER THERE IS A FRONT FRAME IN THE COMPRESSOR

- A- NO FRAME
- B- SNGL BEARING FRAME FOR TFS AND TJS WITHOUT POWER TAKEOFF (PTO)
- C- SINGLE BEARING FRAME WITH PTO
- D- TWO BEARING FRAME WHICH EXTENDS OUTWARD TO THE FAN OUTER CASE AND HOLDS TWO BEARINGS WITH PTO

A  
INDICATE WHETHER THERE IS A REAR FRAME IN THE COMPRESSOR

- A- NO FRAME
- B- SNGL BEARING FRAME FOR TFS AND TJS WITHOUT POWER TAKEOFF (PTO)
- C- SINGLE BEARING FRAME WITH PTO
- D- TWO BEARING FRAME WHICH EXTENDS OUTWARD TO THE FAN OUTER CASE AND HOLDS TWO BEARINGS WITH PTO

D  
GEAR BOX INDICATOR, 0= NO GEAR BOX, N=SHAFT NUMBER FOR GEAR BOX

0  
YOU MAY SPECIFY THE NUMBER OF STAGES-OVERRIDES MAX PR/STG., OTHERWISE ENTER 0

FAN/COMPRESSOR		LOW VALUE-HIGH VALUE	
CODE	VALUE DESCRIPTION		
A	0.5500 FACE INLET MACH NUMBER	0.5000	0.6000
B	1.70000 MAX 1ST STAGE PRATIO	1.50000	1.80000
C	0.450 COMPRESSOR FACE HUB TO TIP RATIO	0.400	0.500
D	1.5000 BLADE SOLIDITY (CORD/SPACING)	1.0000	1.5000
E	4.0000 BLADE ASPECT RATIO-1ST STAGE	3.0000	5.0000
F	3.0000 BLADE ASPECT RATIO-LAST STAGE	2.0000	3.0000
G	0.4500 COMPRESSOR EXIT MACH NUMBER	0.4500	0.5500
H	0.0000 MAX. COMP. INLET T- 0=CALC'D	0.0000	0.0000
I	0.0000 MAX. COMP. EXIT T- 0=CALC'D	0.0000	0.0000
J	1.0000 MAX. SPEED RATIO RPM MAX/RPMD	1.0000	1.0000
K	0.0000 BLADE MATERIAL DENSITY 0=T-SET	0.0000	0.0000
L	2. 1=CONST.HUB,2=MEAN,3=TIP DESH.	0.	0.
M	1.0000 RPM SCALER TO MATCH KNOWN RPM	1.0000	1.0000
N	0.0000 TEMP FOR MATERIAL CHANGE	0.0000	0.0000
O	0.0000 WEIGHT SCALER 0=AS CALCULATED	0.0000	0.0000

P 1.8000 STATOR BLADE TAPER RATIO 0.0000 0.0000  
 S 0.0550 BLADE VOLUME RATIO 0.0000 0.0000  
 ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER  
 Q QUIT PROCESSING ENTIRE ENGINE  
 R REVIEW UPDATED VALUES  
 T TERMINATE - GO ON TO NEXT COMPONENT

B 1.8  
 C .37  
 E 4.5  
 F 4.5  
 G .5

R  
 FAN/COMPRESSOR  

CODE	VALUE	DESCRIPTION	LOW VALUE	HIGH VALUE
A	0.5500	FACE INLET MACH NUMBER	0.5000	0.6000
B	1.80000	MAX 1ST STAGE PRATIO	1.50000	1.80000
C	0.370	COMPRESSOR FACE HUB TO TIP RATIO	0.400	0.500
D	1.5000	BLADE SOLIDITY (CORD/SPACING)	1.0000	1.5000
E	4.5000	BLADE ASPECT RATIO-1ST STAGE	3.0000	5.0000
F	4.5000	BLADE ASPECT RATIO-LAST STAGE	2.0000	3.0000
G	0.5000	COMPRESSOR EXIT MACH NUMBER	0.4500	0.5500
H	0.0000	MAX. COMP. INLET T- 0=CALC'D	0.0000	0.0000
I	0.0000	MAX. COMP. EXIT T- 0=CALC'D	0.0000	0.0000
J	1.0000	MAX. SPEED RATIO RPM MAX/RPM D	1.0000	1.0000
K	0.0000	BLADE MATERIAL DENSITY 0=T-SET	0.0000	0.0000
L	2.	1=CONST.HUB,2=MEAN,3=TIP DESH.	0.	0.
M	1.0000	RPM SCALAR TO MATCH KNOWN RPM	1.0000	1.0000
N	0.0000	TEMP FOR MATERIAL CHANGE	0.0000	0.0000
O	0.0000	WEIGHT SCALAR 0=AS CALCULATED	0.0000	0.0000
P	1.8000	STATOR BLADE TAPER RATIO	0.0000	0.0000
S	0.0550	BLADE VOLUME RATIO	0.0000	0.0000

ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER  
 Q QUIT PROCESSING ENTIRE ENGINE  
 R REVIEW UPDATED VALUES  
 T TERMINATE - GO ON TO NEXT COMPONENT

COMPONENT 3 IS A SPLITTER  
 WEIGHT AND LENGTH ARE IGNORED UNLESS FIRST COMPONENT IN THE ENGINE (AFTER INLET)  
 IF INNER STREAM IS NOT PRIMARY ENTER A "1" OR ELSE JUST HIT RETURN

COMPONENT 6 IS A COMPRESSOR, OPTIONS ARE (ENTER CORRECT LETTER)

- A- TYPICAL FAN
- B- OUTER PORTION OF NON-ROTATING SPLITTER FAN
- C- INNER PORTION OF NON-ROTATING SPLITTER FAN
- D- OUTER PORTION OF ROTATING SPLITTER FAN
- E- INNER PORTION OF ROTATING SPLITTER FAN
- F- LOW PRESSURE COMPRESSOR
- G- HIGH PRESSURE COMPRESSOR

INDICATE WHETHER THERE ARE STATORS OR IF THIS IS A CENTRIFUGAL COMPRESSOR

- A- CALCULATE STATOR WEIGHT
- B- NO STATOR WEIGHT
- C- CENTRIFUGAL COMPRESSOR

INDICATE WHETHER THERE IS A FRONT FRAME IN THE COMPRESSOR

- A- NO FRAME
- B- SNGL BEARING FRAME FOR TFS AND TJS WITHOUT POWER TAKEOFF (PTO)
- C- SINGLE BEARING FRAME WITH PTO
- D- TWO BEARING FRAME WHICH EXTENDS OUTWARD TO THE FAN OUTER CASE AND HOLDS TWO BEARINGS WITH PTO

A  
 INDICATE WHETHER THERE IS A REAR FRAME IN THE COMPRESSOR  
 A- NO FRAME  
 B- SHGL BEARING FRAME FOR TFS AND TJS WITHOUT POWER TAKEOFF (PTO)  
 C- SINGLE BEARING FRAME WITH PTO  
 D- TWO BEARING FRAME WHICH EXTENDS OUTWARD TO THE FAN OUTER CASE AND HOLDS TWO BEARINGS WITH PTO  
 A  
 GEAR BOX INDICATOR, 0= NO GEAR BOX, N=SHAFT NUMBER FOR GEAR BOX

0  
 YOU MAY SPECIFY THE NUMBER OF STAGES-OVERRIDES MAX PR/STG., OTHERWISE ENTER 0

0  
 FAN/COMPRESSOR

CODE	VALUE	DESCRIPTION	LOW VALUE	HIGH VALUE
A	0.5000	FACE INLET MACH NUMBER	0.4500	0.6000
B	1.50000	MAX 1ST STAGE PRATIO	1.50000	1.80000
C	0.400	COMPRESSOR FACE HUB TO TIP RATIO	0.400	0.500
D	1.5000	BLADE SOLIDITY (CORD/SPACING)	1.0000	1.5000
E	4.0000	BLADE ASPECT RATIO-1ST STAGE	3.0000	5.0000
F	3.0000	BLADE ASPECT RATIO-LAST STAGE	2.0000	3.0000
G	0.4500	COMPRESSOR EXIT MACH NUMBER	0.4500	0.5500
H	0.0000	MAX. COMP. INLET T- 0=CALC'D	0.0000	0.0000
I	0.0000	MAX. COMP. EXIT T- 0=CALC'D	0.0000	0.0000
J	1.0000	MAX. SPEED RATIO RPMNAX/RPMD	1.0000	1.0000
K	0.0000	BLADE MATERIAL DENSITY 0=T-SET	0.0000	0.0000
L	2.	1=CONST.HUB,2=MEAN,3=TIP DESN.	0.	0.
M	1.0000	RPM SCALAR TO MATCH KNOWN RPM	1.0000	1.0000
N	0.0000	TEMP FOR MATERIAL CHANGE	0.0000	0.0000
O	0.0000	WEIGHT SCALAR 0=AS CALCULATED	0.0000	0.0000
P	1.2000	STATOR BLADE TAPER RATIO	0.0000	0.0000
S	0.1200	BLADE VOLUME RATIO	0.0000	0.0000

ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER  
 Q QUIT PROCESSING ENTIRE ENGINE  
 R REVIEW UPDATED VALUES  
 T TERMINATE - GO ON TO NEXT COMPONENT

C .85  
 E 2.  
 F 2.  
 L 3.

T  
 COMPONENT 7 IS A DUCT, OPTIONS ARE (ENTER CORRECT LETTER)  
 A- PRIMARY BURNER  
 B- DUCT BURNER  
 C- AUGMENTOR  
 D- DUCT  
 ENTER LETTER

D  
 TYPE OF DUCT  
 A- DUMMY - NO WT OR LENGTH  
 B- INPUT LENGTH ( WILL SPECIFY L/D OF DUCT )  
 C- LENGTH FROM CONNECTION BETWEEN SPLITTER AND MIXER  
 D- CROSSOVER DUCT FOR CENTRIFUGAL COMPRESSORS  
 E- DIFFUSER FOR CENTRIFUGAL COMPRESSORS

B  
 DUCT

CODE	VALUE	DESCRIPTION	LOW VALUE	HIGH VALUE
A	0.40	DUCT MACH NUMBER	0.40	0.50

B 1.00000 LEN/HT IF MODE B 0.00000 0.00000  
 C 0.000 DUCT MEAN DIAM. IF =0 CALC. PER NODE BELOW 0. 0.  
 D -1. IF NODE,0=MEAN D SPECIFIED,-1=CONH. COMP. 0. 0.  
 ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER  
 Q QUIT PROCESSING ENTIRE ENGINE  
 R REVIEW UPDATED VALUES  
 T TERMINATE - GO ON TO NEXT COMPONENT

A .45  
 B 5.5

COMPONENT 8 IS A COMPRESSOR, OPTIONS ARE (ENTER CORRECT LETTER)  
 A- TYPICAL FAN  
 B- OUTER PORTION OF NON-ROTATING SPLITTER FAN  
 C- INNER PORTION OF NON-ROTATING SPLITTER FAN  
 D- OUTER PORTION OF ROTATING SPLITTER FAN  
 E- INNER PORTION OF ROTATING SPLITTER FAN  
 F- LOW PRESSURE COMPRESSOR  
 G- HIGH PRESSURE COMPRESSOR

INDICATE WHETHER THERE ARE STATORS OR IF THIS IS A CENTRIFUGAL COMPRESSOR  
 A- CALCULATE STATOR WEIGHT  
 B- NO STATOR WEIGHT  
 C- CENTRIFUGAL COMPRESSOR  
 A  
 INDICATE WHETHER THERE IS A FRONT FRAME IN THE COMPRESSOR  
 A- NO FRAME  
 B- SNGL BEARING FRAME FOR TFS AND TJS WITHOUT POWER TAKEOFF (PTO)  
 C- SINGLE BEARING FRAME WITH PTO  
 D- TWO BEARING FRAME WHICH EXTENDS OUTWARD TO THE FAN OUTER CASE AND HOLDS TWO BEARINGS WITH PTO  
 A  
 INDICATE WHETHER THERE IS A REAR FRAME IN THE COMPRESSOR  
 A- NO FRAME  
 B- SNGL BEARING FRAME FOR TFS AND TJS WITHOUT POWER TAKEOFF (PTO)  
 C- SINGLE BEARING FRAME WITH PTO  
 D- TWO BEARING FRAME WHICH EXTENDS OUTWARD TO THE FAN OUTER CASE AND HOLDS TWO BEARINGS WITH PTO  
 A

GEAR BOX INDICATOR,0= NO GEAR BOX, N=SHAFT NUMBER FOR GEAR BOX  
 0  
 YOU MAY SPECIFY THE NUMBER OF STAGES-OVERRIDES MAX PR/STG., OTHERWISE ENTER 0  
 0  
 FAN/COMPRESSOR

CODE	VALUE	DESCRIPTION	LOW VALUE	HIGH VALUE
A	0.4000	FACE INLET MACH NUMBER	0.4000	0.5000
B	1.40000	MAX 1ST STAGE PRATIO	1.40000	1.70000
C	0.700	COMPRESSOR FACE HUB TO TIP RATIO	0.600	0.800
D	1.5000	BLADE SOLIDITY (CORD/SPACING)	1.0000	1.5000
E	3.0000	BLADE ASPECT RATIO-1ST STAGE	2.0000	5.0000
F	1.5000	BLADE ASPECT RATIO-LAST STAGE	1.0000	2.0000
G	0.3000	COMPRESSOR EXIT MACH NUMBER	0.2000	0.3000
H	0.0000	MAX. COMP. INLET T- 0=CALC'D	0.0000	0.0000
I	0.0000	MAX. COMP. EXIT T- 0=CALC'D	0.0000	0.0000
J	1.0000	MAX. SPEED RATIO RPM/HPM	1.0000	1.0000
K	0.0000	BLADE MATERIAL DENSITY 0=T-SET	0.0000	0.0000
L	2.	1=CONST.HUB,2=MEAN,3=TIP DESH.	0.	0.
M	1.0000	RPM SCALER TO MATCH KNOWN RPM	1.0000	1.0000
N	0.0000	TEMP FOR MATERIAL CHANGE	0.0000	0.0000
O	0.0000	WEIGHT SCALER 0=AS CALCULATED	0.0000	0.0000
P	1.2000	STATOR BLADE TAPER RATIO	0.0000	0.0000

S 0.1200 BLADE VOLUME RATIO 0.0000 0.0000  
 ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER  
 Q QUIT PROCESSING ENTIRE ENGINE  
 R REVIEW UPDATED VALUES  
 T TERMINATE - GO ON TO NEXT COMPONENT

A .45  
 B 1.31  
 C .47  
 E 4.5  
 L 3.

T  
 COMPONENT 17 IS A DUCT, OPTIONS ARE (ENTER CORRECT LETTER)  
 A- PRIMARY BURNER  
 B- DUCT BURNER  
 C- AUGMENTOR  
 D- DUCT  
 ENTER LETTER

D  
 TYPE OF DUCT  
 A- DUMMY - NO MT OR LENGTH  
 B- INPUT LENGTH ( WILL SPECIFY L/D OF DUCT )  
 C- LENGTH FROM CONNECTION BETWEEN SPLITTER AND MIXER  
 D- CROSSOVER DUCT FOR CENTRIFUGAL COMPRESSORS  
 E- DIFFUSER FOR CENTRIFUGAL COMPRESSORS

B  
 DUCT  

CODE	VALUE	DESCRIPTION	LOW VALUE	HIGH VALUE
A	0.40	DUCT MACH NUMBER	0.40	0.50
B	1.00000	LEN/HT IF NODE B	0.00000	0.00000
C	0.000	DUCT MEAN DIAM. IF =0 CALC. PER NODE BELOW	0.	0.
D	-1.	IF NODE,0=MEAN D SPECIFIED,-1=CONN. COMP.	0.	0.

 ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER  
 Q QUIT PROCESSING ENTIRE ENGINE  
 R REVIEW UPDATED VALUES  
 T TERMINATE - GO ON TO NEXT COMPONENT

A .3  
 B 8.9

T  
 COMPONENT 9 IS A DUCT, OPTIONS ARE (ENTER CORRECT LETTER)  
 A- PRIMARY BURNER  
 B- DUCT BURNER  
 C- AUGMENTOR  
 D- DUCT  
 ENTER LETTER

A  
 A- BURNER HAS A FRAME  
 B- NO FRAME

A  
 PRIMARY BURNER  

CODE	VALUE	DESCRIPTION	LOW VALUE	HIGH VALUE
A	100.00	BURNER THRU-FLOW VELOCITY	100.00	150.00
B	0.01500	AIRFLOW RESIDENCY TIME	0.01000	0.02000
C	0.000	MEAN DIAMETER - IF 0 MATCH* UPSTREAM CMPT.	0.	0.
D	0.	CMPT NUMBER FOR MATCHING DIAMETER	0.	0.
E	0.	NUMBER OF CANS FOR CAN BURNER	0.	0.

 ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER

Q QUIT PROCESSING ENTIRE ENGINE  
 R REVIEW UPDATED VALUES  
 T TERMINATE - GO ON TO NEXT COMPONENT  
 B .018  
 T  
 COMPONENT 10 IS A TURBINE, OPTIONS ARE (ENTER CORRECT LETTER)  
 H- HIGH PRESSURE TURBINE  
 L- LOW PRESSURE TURBINE  
 H  
 A- NO FRAME  
 B- TURBINE HAS EXIT FRAME  
 A  
 ENTER COMPONENT NUMBER OF COMPRESSOR THAT SETS TURBINE RPM  
 08  
 ENTER COMPONENT NUMBER THAT LIMITS TURBINE OUTER RADIUS  
 (+ = OUTLET, - = INLET, 0 = FEEDING COMPONENT)  
 -08  
 YOU MAY ENTER THE NUMBER OF STAGES IF DESIRED, OTHERWISE ENTER 0

0-  
 R- RADIAL FLOW TURBINE  
 A- AXIAL FLOW TURBINE  
 A  
 TURBINE  

CODE	VALUE	DESCRIPTION	LOW VALUE	HIGH VALUE
A	0.3000	FACE INLET MACH NUMBER	0.3000	0.4000
B	0.28000	LOADING PARAMETER	0.20000	0.30000
C	1.500	BLADE SOLIDITY (CORD/SPACING)	1.000	1.500
D	1.5000	BLADE ASPECT RATIO- 1ST STAGE	1.0000	2.0000
E	1.5000	BLADE ASPECT RATIO-LAST STAGE	1.0000	2.0000
F	0.4500	TURBINE EXIT MACH NUMBER	0.4500	0.5000
G	125000.	DISC REFERENCE STRESS	100000.	150000.
H	2.	1=CONST TIP DESH, 2=MEAN, 3=HUB	0.	0.
I	1.0000	MAX. SPEED RATIO RPM/AX/RPMD	1.0000	1.0000
J	0.0000	CONTROL RADIUS- 0 IF TRANSFRD	0.0000	0.0000
K	0.0000	BLADE MATERIAL DENSITY 0=T-SET	0.0000	0.0000
L	0.1550	BLADE VOLUME FACTOR	0.0000	0.0000
P	1.0000	BLADE TAPER RATIO	0.0000	0.0000
S	0.1550	STATOR BLADE VOLUME FACTOR	0.0000	0.0000

ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER  
 Q QUIT PROCESSING ENTIRE ENGINE  
 R REVIEW UPDATED VALUES  
 T TERMINATE - GO ON TO NEXT COMPONENT

A .4  
 B .2  
 F .5  
 G 150000.  
 H 3.

T  
 COMPONENT 11 IS A DUCT, OPTIONS ARE (ENTER CORRECT LETTER)  
 A- PRIMARY BURNER  
 B- DUCT BURNER  
 C- AUGMENTOR  
 D- DUCT  
 ENTER LETTER

D  
 TYPE OF DUCT  
 A- DUMMY - NO WT OR LENGTH

B- INPUT LENGTH ( WILL SPECIFY L/D OF DUCT )  
 C- LENGTH FROM CONNECTION BETWEEN SPLITTER AND MIXER  
 D- CROSSOVER DUCT FOR CENTRIFUGAL COMPRESSORS  
 E- DIFFUSER FOR CENTRIFUGAL COMPRESSORS

B  
 DUCT  

CODE	VALUE	DESCRIPTION	LOW VALUE-HIGH VALUE	
A	0.40	DUCT MACH NUMBER	0.40	0.50
B	1.00000	LEN/HT IF NODE B	0.00000	0.00000
C	0.000	DUCT MEAN DIAM. IF =0 CALC. PER NODE BELOW	0.	0.
D	-1.	IF NODE,0=MEAN D SPECIFIED,-1=CONN. COMP.	0.	0.

 ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER  
 Q QUIT PROCESSING ENTIRE ENGINE  
 R REVIEW UPDATED VALUES  
 T TERMINATE - GO ON TO NEXT COMPONENT

A .5  
 B 3.5

T  
 COMPONENT 12 IS A TURBINE, OPTIONS ARE (ENTER CORRECT LETTER)

H- HIGH PRESSURE TURBINE

L- LOW PRESSURE TURBINE

L

A- NO FRAME

B- TURBINE HAS EXIT FRAME

B

ENTER COMPONENT NUMBER OF COMPRESSOR THAT SETS TURBINE RPM

02

ENTER COMPONENT NUMBER THAT LIMITS TURBINE OUTER RADIUS  
 (+ = OUTLET, - = INLET, 0 = FEEDING COMPONENT)

436

YOU MAY ENTER THE NUMBER OF STAGES IF DESIRED, OTHERWISE ENTER 0

0

R- RADIAL FLOW TURBINE

A- AXIAL FLOW TURBINE

A

TURBINE

CODE	VALUE	DESCRIPTION	LOW VALUE-HIGH VALUE	
A	0.4500	FACE INLET MACH NUMBER	0.4000	0.5000
B	0.28000	LOADING PARAMETER	0.20000	0.30000
C	1.500	BLADE SOLIDITY (CORD/SPACING)	1.000	1.500
D	2.0000	BLADE ASPECT RATIO- 1ST STAGE	2.0000	3.0000
E	4.0000	BLADE ASPECT RATIO-LAST STAGE	4.0000	6.0000
F	0.5500	TURBINE EXIT MACH NUMBER	0.5500	0.6000
G	125000.	DISC REFERENCE STRESS	100000.	150000.
H	2.	1=CONST TIP DESN,2=MEAN,3=HUB	0.	0.
I	1.0000	MAX. SPEED RATIO RPM/HPD	1.0000	1.0000
J	0.0000	CONTROL RADIUS- 0 IF TRANSFRD	0.0000	0.0000
K	0.0000	BLADE MATERIAL DENSITY 0-T-SET	0.0000	0.0000
L	0.1950	BLADE VOLUME FACTOR	0.0000	0.0000
P	1.0000	BLADE TAPER RATIO	0.0000	0.0000
S	0.1950	STATOR BLADE VOLUME FACTOR	0.0000	0.0000

ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER

Q QUIT PROCESSING ENTIRE ENGINE

R REVIEW UPDATED VALUES

T TERMINATE - GO ON TO NEXT COMPONENT

A .5  
 B .3



F .6  
G 150000.  
H 1.  
T  
COMPONENT 13 IS A DUCT, OPTIONS ARE (ENTER CORRECT LETTER)  
A- PRIMARY BURNER  
B- DUCT BURNER  
C- AUGMENTOR  
D- DUCT  
ENTER LETTER

D  
TYPE OF DUCT  
A- DUMMY - NO WT OR LENGTH  
B- INPUT LENGTH ( WILL SPECIFY L/D OF DUCT )  
C- LENGTH FROM CONNECTION BETWEEN SPLITTER AND MIXER  
D- CROSSOVER DUCT FOR CENTRIFUGAL COMPRESSORS  
E- DIFFUSER FOR CENTRIFUGAL COMPRESSORS

A  
DUCT  
CODE VALUE DESCRIPTION LOW VALUE-HIGH VALUE  
A 0.40 DUCT MACH NUMBER 0.40 0.50  
B 1.00000 LEN/HT IF MODE B 0.00000 0.00000  
C 0.000 DUCT MEAN DIAM. IF =0 CALC. PER NODE BELOW 0. 0.  
D -1. IF NODE,0=MEAN D SPECIFIED,-1=CONN. COMP. 0. 0.  
ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER  
Q QUIT PROCESSING ENTIRE ENGINE  
R REVIEW UPDATED VALUES  
T TERMINATE - GO ON TO NEXT COMPONENT

A .6  
B .5  
T  
COMPONENT 14 IS A NOZZLE, ENTER NOZZLE TYPE  
1= CONVERGENT  
2= C-D VARIABLE AREA  
1  
ENTER COMPONENT NUMBER THAT LIMITS NOZZLE OUTER RADIUS  
(+ = OUTLET, - = INLET, 0 = FEEDING COMPONENT)  
0  
ENTER THRUST REVERSER TYPE, 0=NONE, 1=FAN, 2=PRIMARY

0  
NOZZLE  
CODE VALUE DESCRIPTION LOW VALUE-HIGH VALUE  
A 1.0000 LENGTH TO DIAMETER RATIO 1.0000 2.0000  
B 0.0000 BYPASS RATIO FOR MIXED FLOW 0.0000 0.0000  
ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER  
Q QUIT PROCESSING ENTIRE ENGINE  
R REVIEW UPDATED VALUES  
T TERMINATE - GO ON TO NEXT COMPONENT

T  
COMPONENT 4 IS A DUCT, OPTIONS ARE (ENTER CORRECT LETTER)  
A- PRIMARY BURNER  
B- DUCT BURNER  
C- AUGMENTOR  
D- DUCT  
ENTER LETTER

D

TYPE OF DUCT  
 A- DUMMY - NO WT OR LENGTH  
 B- INPUT LENGTH ( WILL SPECIFY L/D OF DUCT )  
 C- LENGTH FROM CONNECTION BETWEEN SPLITTER AND MIXER  
 D- CROSSOVER DUCT FOR CENTRIFUGAL COMPRESSORS  
 E- DIFFUSER FOR CENTRIFUGAL COMPRESSORS

B  
 DUCT  

CODE	VALUE	DESCRIPTION	LOW VALUE	HIGH VALUE
A	0.40	DUCT MACH NUMBER	0.40	0.50
B	1.00000	LEN/HT IF MODE B	0.00000	0.00000
C	0.000	DUCT MEAN DIAM. IF =0 CALC. PER NODE BELOW	0.	0.
D	-1.	IF NODE,0=MEAN D SPECIFIED,-1=CONN. COMP.	0.	0.

  
 ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER  
 Q QUIT PROCESSING ENTIRE ENGINE  
 R REVIEW UPDATED VALUES  
 T TERMINATE - GO ON TO NEXT COMPONENT

A .5  
 T  
 COMPONENT 5 IS A NOZZLE, ENTER NOZZLE TYPE  
 1= CONVERGENT  
 2= C-D VARIABLE AREA  
 1  
 ENTER COMPONENT NUMBER THAT LIMITS NOZZLE OUTER RADIUS  
 (+ = OUTLET, - = INLET, 0 = FEEDING COMPONENT)  
 0  
 ENTER THRUST REVERSER TYPE, 0=NONE, 1=FAN, 2=PRIMARY

0  
 NOZZLE  

CODE	VALUE	DESCRIPTION	LOW VALUE	HIGH VALUE
A	1.0000	LENGTH TO DIAMETER RATIO	1.0000	2.0000
B	0.0000	BYPASS RATIO FOR MIXED FLOW	0.0000	0.0000

  
 ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER  
 Q QUIT PROCESSING ENTIRE ENGINE  
 R REVIEW UPDATED VALUES  
 T TERMINATE - GO ON TO NEXT COMPONENT

T  
 COMPONENT 15 IS A SHAFT, ENTER SHAFT NUMBER FROM INNER TO OUTER, I.E. 1 2 3 ETC.

T  
 ENTER TURBINES CONNECTED TO THIS SHAFT

12  
 ENTER COMPONENT NUMBER OF FIRST COMPRESSOR ON THIS SHAFT

02  
 SHAFT  

CODE	VALUE	DESCRIPTION	LOW VALUE	HIGH VALUE
A	50000.00	SHAFT ALLOWABLE STRESS	40000.00	50000.00
B	0.29	SHAFT MATERIAL DENSITY	0.28	0.31
C	0.0000	DIAMETER RATIO (INNER/OUTER)	0.0000	0.8500
D	0.	COMPONENT NUMBER ON SHAFT FOR INERTIA	0.	0.
E	0.	COMPONENT NUMBER ON SHAFT FOR INERTIA	0.	0.
F	0.	COMPONENT NUMBER ON SHAFT FOR INERTIA	0.	0.
G	0.	COMPONENT NUMBER ON SHAFT FOR INERTIA	0.	0.

ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER  
 Q QUIT PROCESSING ENTIRE ENGINE

R REVIEW UPDATED VALUES  
T TERMINATE - GO ON TO NEXT COMPONENT

B .3  
C .85  
D 2.  
E 6.  
F 12.

T COMPONENT 16 IS A SHAFT, ENTER SHAFT NUMBER FROM INNER TO OUTER, I.E. 1 2 3 ETC.

2  
ENTER TURBINES CONNECTED TO THIS SHAFT

10  
ENTER COMPONENT NUMBER OF FIRST COMPRESSOR ON THIS SHAFT

SHAFT		LOW VALUE-HIGH VALUE	
CODE	VALUE DESCRIPTION		
A	50000.00 SHAFT ALLOWABLE STRESS	40000.00	50000.00
B	0.29 SHAFT MATERIAL DENSITY	0.28	0.31
C	0.0000 DIAMETER RATIO (INNER/OUTER)	0.0000	0.8500
D	0. COMPONENT NUMBER ON SHAFT FOR INERTIA	0.	0.
E	0. COMPONENT NUMBER ON SHAFT FOR INERTIA	0.	0.
F	0. COMPONENT NUMBER ON SHAFT FOR INERTIA	0.	0.
G	0. COMPONENT NUMBER ON SHAFT FOR INERTIA	0.	0.

ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER

Q QUIT PROCESSING ENTIRE ENGINE  
R REVIEW UPDATED VALUES  
T TERMINATE - GO ON TO NEXT COMPONENT

B .3  
D 8.  
E 10.

T TERMINATED: STOP

# APPENDIX B

Card Images from PREPWATE AND NNEP/WATE Input Dataset

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&N IPLT= T,PLOT= T,ISII= F,ISIO= F,
INT=4,IOUTCD=2,
DISKWI=0.,ACCS=0.1,ISCALE(1)=2.3,SCALE(1)=1.,.8,1.2,ACCARM=0,
ILENG(1)= 2, 3, 6, 7, 8,17, 9,10,11,12,13,14,
IWMEC(1, 2)=4HFAN , 0, 0, 4, 0, 0, 0,
IWMEC(1, 3)=4HSPLT, 0, 0, 0, 0, 0, 0,
IWMEC(1, 4)=4HDUCT, 2, 0, 0, 0, 0, 0,
IWMEC(1, 5)=4HNOZ , 1, 0, 0, 0, 0, 0,
IWMEC(1, 6)=4HLLPC , 1, 0, 0, 0, 0, 0,
IWMEC(1, 7)=4HDUCT, 2, 0, 0, 0, 0, 0,
IWMEC(1, 8)=4HHPC , 1, 0, 0, 0, 0, 0,
IWMEC(1, 9)=4HPBUR, 1, 0, 0, 0, 0, 0,
IWMEC(1,10)=4HHPT , 0, 8, -8, 0, 0, 0,
IWMEC(1,11)=4HDUCT, 2, 0, 0, 0, 0, 0,
IWMEC(1,12)=4HLPT , 1, 2, 6, 0, 0, 0,
IWMEC(1,13)=4HDUCT, 1, 0, 0, 0, 0, 0,
IWMEC(1,14)=4HNOZ , 1, 0, 0, 0, 0, 0,
IWMEC(1,15)=4HSHAF, 1, 12, 0, 0, 0, 2,
IWMEC(1,16)=4HSHAF, 2, 10, 0, 0, 0, 8,
IWMEC(1,17)=4HDUCT, 2, 0, 0, 0, 0, 0,
DESVAL(1,2)=0.5500,1.800,0.3700,1.500,4.500,0.5000,
0.0000,0.0000,1.000,0.0000,2.000,1.000,0.0000,0.0000,1.800,0.5500D-01,
DESVAL(1,3)=0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,4)=0.5000,1.000,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,5)=1.000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,6)=0.5000,1.500,0.8500,1.500,2.000,2.000,0.4500,
0.0000,0.0000,1.000,0.0000,3.000,1.000,0.0000,0.0000,1.200,0.1200,
DESVAL(1,7)=0.4500,5.500,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,8)=0.4500,1.310,0.4700,1.500,4.500,1.500,0.3000,
0.0000,0.0000,1.000,0.0000,3.000,1.000,0.0000,0.0000,1.200,0.1200,
DESVAL(1,9)=100.0,0.1800D-01,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,10)=0.4000,0.2000,1.500,1.500,1.500,0.5000,0.1500D06,
3.000,1.000,0.0000,0.0000,0.1550,0.0000,0.0000,0.0000,1.000,0.1550,
DESVAL(1,11)=0.5000,3.500,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,12)=0.5000,0.3000,1.500,2.000,4.000,0.6000,0.1500D06,
1.000,1.000,0.0000,0.0000,0.1950,0.0000,0.0000,0.0000,1.000,0.1950,
DESVAL(1,13)=0.6000,0.5000,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,14)=1.000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,15)=0.5000D05,0.3000,0.8500,2.000,6.000,12.00,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,16)=0.5000D05,0.3000,0.0000,8.000,10.00,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,17)=0.3000,8.900,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
&END

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SAMPLE CASE FOR PREPWATE

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&D
&END
&D MODE=1,
IWT=4,
KONFIG(1,1)='INLT',1,0,2,0,SPEC(1,1)=1000.,4*0.1.,2*0..00001,2*0,20.,
KONFIG(1,2)='COMP',2,0,3,0,SPEC(1,2)=1.8,0,1,3761,1,3762,1,3763,1,0,0..824,
1.734,1,
KONFIG(1,3)='SPLY',3,0,7,4,SPEC(1,3)=4.323,
KONFIG(1,4)='DUCT',4,0,5,0,SPEC(1,4)=6*0,
KONFIG(1,5)='NOZZ',5,0,6,0,SPEC(1,5)=0,1,0,0.,99,0,0,0,1,
KONFIG(1,6)='CO'IP',7,0,8,0,SPEC(1,6)=1.3,0,1,3761,1,3762,1,3763,1,0,0..867,
1.373,1,
KONFIG(1,7)='DUCT',8,0,9,0,SPEC(1,7)=6*0,
KONFIG(1,8)='COMP',9,0,10,17,SPEC(1,8)=1.3..162,1,3707,1,3708,1,3709,1,0,0,
.867,12,75..985,
KONFIG(1,9)='DUCT',19,0,11,0,SPEC(1,9)=.048,0,0,2990..1..18400.,
KONFIG(1,10)='TURB',11,17,12,0,SPEC(1,10)=4..79,1,3801,1,3802,1,1..625,
1..919,5650,1,
KONFIG(1,11)='DUCT',12,0,13,0,SPEC(1,11)=6*0,
KONFIG(1,12)='TURB',13,17,14,0,SPEC(1,12)=2.5..21,1,3803,1,3804,1,1,0,1..912,
5244,1,
KONFIG(1,13)='DUCT',14,0,15,0,SPEC(1,13)=6*0,
KONFIG(1,14)='NOZZ',15,0,16,0,SPEC(1,14)=0,1,0,0..99,0,0,0,1,
KONFIG(1,15)='SHFT',2,6,12,0,SPEC(1,15)=4000,8*1,
KONFIG(1,16)='SHFT',8,10,0,0,SPEC(1,16)=6000,8*1,
KONFIG(1,17)='DUCT',10,0,19,0,SPEC(1,17)=6*0,
&END
IN IPLT= T, PLOT= T, ISII= F, ISIO= F,
IWT=4, IOUTCD=2,
DISKWI=0., ACC5=0.1, ISCALE(1)=2,3,SCALE(1)=1..8,1.2, ACCARM=0,
ILEHG(1)= 2, 3, 6, 7, 8,17, 9,10,11,12,13,14,
IMMEC(1, 2)=4HFAN , 0, 0, 4, 0, 0, 0,
IMMEC(1, 3)=4HSPLY, 0, 0, 0, 0, 0, 0,
IMMEC(1, 4)=4HDUCT, 2, 0, 0, 0, 0, 0,
IMMEC(1, 5)=4HNOZ , 1, 0, 0, 0, 0, 0,
IMMEC(1, 6)=4HLPFC, 1, 0, 0, 0, 0, 0,
IMMEC(1, 7)=4HDUCT, 2, 0, 0, 0, 0, 0,
IMMEC(1, 8)=4HIPC , 1, 0, 0, 0, 0, 0,
IMMEC(1, 9)=4HPBUR, 1, 0, 0, 0, 0, 0,
IMMEC(1,10)=4HHPT , 0, 8, -8, 0, 0, 0,
IMMEC(1,11)=4HDUCT, 2, 0, 0, 0, 0, 0,
IMMEC(1,12)=4HLPT , 1, 2, 6, 0, 0, 0,
IMMEC(1,13)=4HDUCT, 1, 0, 0, 0, 0, 0,
IMMEC(1,14)=4HNOZ , 1, 0, 0, 0, 0, 0,
IMMEC(1,15)=4HSHAF, 1, 12, 0, 0, 0, 2,
IMMEC(1,16)=4HSHAF, 2, 10, 0, 0, 0, 8,
IMMEC(1,17)=4HDUCT, 2, 0, 0, 0, 0, 0,
DESVAL(1,2)=0.5500,1.800,0.3700,1.500,4.500,4.500,0.5000,
0.0000,0.0000,1.000,0.0000,2.000,1.000,0.0000,0.0000,1.800,0.5500D-01,
DESVAL(1,3)=0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,4)=0.5000,1.000,0.0000,-1.000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,5)=1.000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,6)=0.5000,1.500,0.8500,1.500,2.000,2.000,0.4500,
0.0000,0.0000,1.000,0.0000,3.000,1.000,0.0000,0.0000,1.200,0.1200,
DESVAL(1,7)=0.4500,5.500,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,

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DESVAL(1,8)=0.4500,1.310,0.4700,1.500,4.500,1.500,0.3000,
0.0000,0.0000,1.000,0.0000,3.000,1.000,0.0000,0.0000,1.200,0.1200,
DESVAL(1,9)=100.0,0.1800D-01,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,10)=0.4000,0.2000,1.500,1.500,1.500,0.5000,0.1500D06,
3.000,1.000,0.0000,0.0000,0.1550,0.0000,0.0000,0.0000,1.000,0.1550,
DESVAL(1,11)=0.5000,3.500,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,12)=0.5000,0.3000,1.500,2.000,4.000,0.6000,0.1500D06,
1.000,1.000,0.0000,0.0000,0.1950,0.0000,0.0000,0.0000,1.000,0.1950,
DESVAL(1,13)=0.6000,0.5000,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,14)=1.000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,15)=0.5000D05,0.3000,0.8500,2.000,6.000,12.00,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,16)=0.5000D05,0.3000,0.0000,8.000,10.00,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,17)=0.3000,8.900,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
&END

```

## APPENDIX C

Output from Sample Case from NNEP/WATE Code

```
SAMPLE CASE FOR PREPWATE  
&D  
&END
```

## TABLE DATA INPUT SUMMARY 16 TABLES

TABLE NUMBER	REFERENCE NUMBER	ARRAY LOCATION
1	3761	1
2	3762	1075
3	3763	2149
4	3764	3223
5	3765	4297
6	3766	5371
7	3767	6445
8	3768	7681
9	3769	8917
10	3801	10153
11	3802	10606
12	3803	11203
13	3804	11656
14	3901	12397
15	3902	12799
16	3903	13213

DATA STORAGE ALLOCATION 20000  
DATA STORAGE NOT USED 6385

```

&D MODE=1,
IWT=4,
KONFIG(1,1)='INLT',1,0,2,0,SPEC(1,1)=1000.,4*0,1.,2*0.,.00001,2*0,20.,
KONFIG(1,2)='COMP',2,0,3,0,SPEC(1,2)=1.8,0,1,3761,1,3762,1,3763,1,0,0.,.824,
1.734,1,
KONFIG(1,3)='SPLIT',3,0,7,4,SPEC(1,3)=4.323,
KONFIG(1,4)='DUCT',4,0,5,0,SPEC(1,4)=6*0,
KONFIG(1,5)='HOZZ',5,0,6,0,SPEC(1,5)=0,1,0,0.,.99,0,0,0,1,
KONFIG(1,6)='COMP',7,0,8,0,SPEC(1,6)=1.3,0,1,3761,1,3762,1,3763,1,0,0.,.867,
1.373,1,
KONFIG(1,7)='DUCT',8,0,9,0,SPEC(1,7)=6*0,
KONFIG(1,8)='COMP',9,0,10,17,SPEC(1,8)=1.3.,.162,1,3707,1,3708,1,3709,1,0,0,
.867,12.75.,.985,
KONFIG(1,9)='DUCT',19,0,11,0,SPEC(1,9)=.048,0,0,2990.,1.,.18400.,
KONFIG(1,10)='TURB',11,17,12,0,SPEC(1,10)=4.,.79,1,3801,1,3802,1,1.,.625,
1.,.919,5680,1,
KONFIG(1,11)='DUCT',12,0,13,0,SPEC(1,11)=6*0,
KONFIG(1,12)='TURB',13,17,14,0,SPEC(1,12)=2.5.,.21,1,3803,1,3804,1,1,0,1.,.912,
5244,1,
KONFIG(1,13)='DUCT',14,0,15,0,SPEC(1,13)=6*0,
KONFIG(1,14)='HOZZ',15,0,16,0,SPEC(1,14)=0,1,0,0.,.99,0,0,0,1,
KONFIG(1,15)='SHFT',2,6,12,0,SPEC(1,15)=4000,8*1,
KONFIG(1,16)='SHFT',8,10,0,0,SPEC(1,16)=6000,8*1,
KONFIG(1,17)='DUCT',10,0,19,0,SPEC(1,17)=6*0,
&END

```

THE FOLLOWING REPRESENTS THE CONFIGURATION FOR MODE= 1  
SAMPLE CASE FOR PREPWATE  
CONFIGURATION DATA 19 STATIONS 17 COMPONENTS

COMPONENT NUMBER	NKIND	COMPONENT TYPE	UPSTREAM STATIONS	DOWNSTREAM STATIONS
1	1	INLET	1	0
2	4	COMPRESR	2	0
3	7	SPLITTER	3	0



4	2	DUCT B	4	0	5	0
5	9	NOZZLE	5	0	6	0
6	4	COMPRESR	7	0	8	0
7	2	DUCT B	8	0	9	0
8	4	COMPRESR	9	0	10	17
9	2	DUCT B	19	0	11	0
10	5	TURBINE	11	17	12	0
11	2	DUCT B	12	0	13	0
12	5	TURBINE	13	17	14	0
13	2	DUCT B	14	0	15	0
14	9	NOZZLE	15	0	16	0
15	11	SHAFT	2	6	12	0
16	11	SHAFT	8	10	0	0
17	2	DUCT B	10	0	19	0

CASE IDENTIFICATION      SAMPLE CASE FOR PREPWATE

# INPUT DATA

COMPONENT NO.	TYPE	DATINP1	DATINP2	DATINP3	DATINP4	DATINP5	DATINP6	DATINP7	DATINP8	DATINP9
1	INLET	0.10000D 04	0.00000	0.00000	0.00000	0.00000	0.10000D 01	0.00000	0.00000	0.10000D-04
2	COMPRESR	0.18000D 01	0.00000	0.10000D 01	0.37610D 04	0.10000D 01	0.37620D 04	0.10000D 01	0.37630D 04	0.10000D 01
3	SPLITTER	0.43230D 01	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
4	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
5	NOZZLE	0.00000	0.10000D 01	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
6	COMPRESR	0.13000D 01	0.00000	0.10000D 01	0.37610D 04	0.10000D 01	0.37620D 04	0.10000D 01	0.37630D 04	0.10000D 01
7	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
8	COMPRESR	0.13000D 01	0.16200D 00	0.10000D 01	0.37070D 04	0.10000D 01	0.37080D 04	0.10000D 01	0.37090D 04	0.10000D 01
9	DUCT B	0.48000D-01	0.00000	0.00000	0.29900D 04	0.10000D 01	0.18400D 05	0.00000	0.00000	0.00000
10	TURBINE	0.40000D 01	0.79000D 00	0.10000D 01	0.38010D 04	0.10000D 01	0.38020D 04	0.10000D 01	0.38030D 04	0.62500D 00
11	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
12	TURBINE	0.25000D 01	0.21000D 00	0.10000D 01	0.38030D 04	0.10000D 01	0.38040D 04	0.10000D 01	0.38050D 04	0.00000
13	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
14	NOZZLE	0.00000	0.10000D 01	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
15	SHAFT	0.40000D 04	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01
16	SHAFT	0.60000D 04	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01
17	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000

```

MODE      1 NOW BEING USED
SUM OF (ERRORS**2)=      0.00000
4W IPLT= T,PLOT= T,ISII= F,ISIO= F,
IWT=4,IOUTCD=2,
DISKWI=0.,ACCS=0.1,ISCALE(1)=2.3,SCALE(1)=1...8,1.2,ACCARM=0,
ILENG(1)= 2, 3, 6, 7, 8,17, 9,10,11,12,13,14,
IWMEC(1, 2)=4HFAN , 0, 0, 4, 0, 0, 0,
IWMEC(1, 3)=4HSPLT, 0, 0, 0, 0, 0, 0,
IWMEC(1, 4)=4HDUCT, 2, 0, 0, 0, 0, 0,
IWMEC(1, 5)=4HNOZ , 1, 0, 0, 0, 0, 0,
IWMEC(1, 6)=4HLPC , 1, 0, 0, 0, 0, 0,
IWMEC(1, 7)=4HDUCT, 2, 0, 0, 0, 0, 0,
IWMEC(1, 8)=4HHPC , 1, 0, 0, 0, 0, 0,
IWMEC(1, 9)=4HPBUR, 1, 0, 0, 0, 0, 0,
IWMEC(1,10)=4HHPT , 0, 8, -8, 0, 0, 0,
IWMEC(1,11)=4HDUCT, 2, 0, 0, 0, 0, 0,
IWMEC(1,12)=4HLPT , 1, 2, 6, 0, 0, 0,
IWMEC(1,13)=4HDUCT, 1, 0, 0, 0, 0, 0,
IWMEC(1,14)=4HNOZ , 1, 0, 0, 0, 0, 0,
IWMEC(1,15)=4HSHAF, 1, 12, 0, 0, 0, 2,
IWMEC(1,16)=4HSHAF, 2, 10, 0, 0, 0, 8,
IWMEC(1,17)=4HDUCT, 2, 0, 0, 0, 0, 0,
DESVAL(1,2)=0.5500,1.800,0.3700,1.500,4.500,4.500,0.5000,
0.0000,0.0000,1.000,0.0000,2.000,1.000,0.0000,0.0000,1.800,0.5500D-01,
DESVAL(1,3)=0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,4)=0.5000,1.000,0.0000,-1.000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,5)=1.000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,6)=0.5000,1.500,0.8500,1.500,2.000,2.000,0.4500,
0.0000,0.0000,1.000,0.0000,3.000,1.000,0.0000,0.0000,1.200,0.1200,
DESVAL(1,7)=0.4500,5.500,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,8)=0.4500,1.310,0.4700,1.500,4.500,1.500,0.3000,
0.0000,0.0000,1.000,0.0000,3.000,1.000,0.0000,0.0000,1.200,0.1200,
DESVAL(1,9)=100.0,0.1800D-01,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,10)=0.4000,0.2000,1.500,1.500,1.500,0.5000,0.1500D06,
3.000,1.000,0.0000,0.0000,0.1550,0.0000,0.0000,0.0000,1.000,0.1550,
DESVAL(1,11)=0.5000,3.500,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,12)=0.5000,0.3000,1.500,2.000,4.000,0.6000,0.1500D06,
1.000,1.000,0.0000,0.0000,0.1950,0.0000,0.0000,0.0000,1.000,0.1950,
DESVAL(1,13)=0.6000,0.5000,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,14)=1.000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,15)=0.5000D05,0.3000,0.8500,2.000,6.000,12.00,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,16)=0.5000D05,0.3000,0.0000,8.000,10.00,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,17)=0.3000,8.900,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
--END

```

```

*****
X      X

```

\* FAN 2 \*  
\*  
\*\*\*\*\*2

MAX CONDITIONS OCCUR AT

\*\*\*\*\*

	ALT	MH	VALUE
PTOT	0.	0.000	14.7 LB/SQIN
TTOT	0.	0.000	538.7 DEG R
CWIN	0.	0.000	1019.1 LB/SEC

\*\*\*\*\*

DUCT

M NO	VEL	T TOT	P TOT	P STAT	AREA	GAM
0.550	608.	539.	2116.	1723.	25.8790	1.4002

UTIPMAX STRESS	DEN	W/AREA	TR	H/T
1591.8	45894.5	0.168	4.970	1.800

COMPRESSOR 2 MECHANICAL DESIGN

LOADING	N STG	DIAM	U TIP C	RPM	C RPM	MAX RPM
0.927	1.00	74.14	1562.0	4920.4	4828.2	4920.4

STAGE 1

WD	WB	WS	WN	WC	CL	RHOB	RHOD	AR
225.	389.	0.	0.	35.	5.2	0.168	0.168	4.50

PR	DEL H	MACH	AREA	R HUB	R TIP	NB	UTIPMAX	STR	WEIGHT	TIN	TMAX	STAGE I
1.7337	26.7	0.550	25.879	13.72	37.07	67	1591.8	45894.	649.	539.	539.	165288.

FRAME WT = 806.03

N STG	WEIGHT	LENGTH	CENGRA	INERTIA
1	1455.30	10.38	8.1	165287.8

DUCT

M NO	VEL	T TOT	P TOT	P STAT	AREA	GAM
0.550	667.	650.	3670.	2989.	16.4039	1.3977

PR	AD EF	PD	TO	HP
1.7340	0.8240	3669.5	649.8	37781.

HI	HO	WI	CWI
128.75	155.45	1000.00	1019.09

\*\*\*\*\* TOTAL COMP WEIGHT IS 1455.303

\*\*\*\*\*  
\*  
\* LPC 6 \*  
\*  
\*\*\*\*\*2

MAX CONDITIONS OCCUR AT

\*\*\*\*\*

	ALT	MH	VALUE
PTOT	0.	0.000	25.5 LB/SQIN
TTOT	0.	0.000	649.8 DEG R
CWIN	0.	0.000	121.3 LB/SEC

\*\*\*\*\*

DUCT  
M NO VEL T TOT P TOT P STAT AREA GAM  
0.500 609. 650. 3670. 3094. 3.2903 1.3977

UTIPMAX STRESS DEN W/AREA TR H/T  
1001.0 7725.8 0.168 0.627 1.200 0.850

# COMPRESSOR 6 MECHANICAL DESIGN

LOADING N STG DIAM U TIP C RPM C RPM MAX RPM  
0.866 1.00 46.63 894.3 4920.4 4396.1 4920.4

STAGE 1  
WD WB WS WN WC CL RHOB RHOD AR  
71. 31. 31. 162. 17. 4.1 0.168 0.168 2.00  
PR DEL H MACH AREA R HUB R TIP HB UTIPMAX STR WEIGHT TIN TMAX STAGE I  
1.3741 17.1 0.500 3.290 19.82 23.31 125 1001.0 7726. 314. 650. 650. 26483.

N STG WEIGHT LENGTH CENGRA INERTIA  
1 313.76 4.09 4.1 26482.8

DUCT  
M NO VEL T TOT P TOT P STAT AREA GAM  
0.500 641. 721. 5038. 4250. 2.5254 1.3953

PR AD EF PO TO HP  
1.3730 0.8670 5038.3 720.6 4542.  
HI HO WI CWI  
155.45 172.54 187.86 121.26

\*\*\*\*\* TOTAL COMP WEIGHT IS 313.764

\*\*\*\*\*  
\* \*  
\* DUCT 7 \*  
\* \*  
\*\*\*\*\*2

## MAX CONDITIONS OCCUR AT

\*\*\*\*\*  
ALT MH  
PTOT 0. 0.000  
TTOT 0. 0.000  
\*\*\*\*\*

DUCT , 7  
RH= 20.40 RT= 23.27 LENG= 15.76  
AREA= 2.731 RHOD=.168  
CAS WT INC WT WTOT  
6.3043 5.5279 11.8322

\*\*\*\*\*  
\* \*  
\* HPC 8 \*  
\* \*  
\*\*\*\*\*2

## MAX CONDITIONS OCCUR AT

\*\*\*\*\*

ALT      MN      VALUE  
PTOT    0.      0.000    35.0 LB/SQIN  
TTOT    0.      0.000    720.6 DEG R  
CWIN    0.      0.000    93.0 LB/SEC  
XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX  
DUCT  
M NO VEL T TOT P TOT P STAT AREA GAM  
0.450 580. 721. 5038. 4387. 2.7306 1.3953

UTIPMAX STRESS    DEN W/AREA    YR    H/T  
1198.8 30837.8    0.168    1.204    1.200    0.470

# COMPRESSOR & MECHANICAL DESIGN

LOADING    N STG    DIAM    U TIP C    RPM    C RPM    MAX RPM  
0.842    14.08    25.35    1017.1    10838.5    9195.5    10838.5

STAGE    1  
WD    WB    WS    WN    WC    CL    RHOB    RHOD    AR  
9.    24.    24.    13.    8.    3.5    0.168    0.168    4.50  
PR DEL H MACH AREA R HUB R TIP NB UTIPMAX STR WEIGHT TIN TMAX STAGE I  
1.3028 15.1 0.450 2.731 5.96 12.67 80 1198.8 30838. 77. 721. 721. 1372.

STAGE    2  
WD    WB    WS    WN    WC    CL    RHOB    RHOD    AR  
15.    14.    14.    10.    6.    2.8    0.168    0.168    4.27  
PR DEL H MACH AREA R HUB R TIP NB UTIPMAX STR WEIGHT TIN TMAX STAGE I  
1.2768 15.1 0.439 2.227 7.65 12.67 101 1198.8 25244. 60. 783. 783. 1397.

STAGE    3  
WD    WB    WS    WN    WC    CL    RHOB    RHOD    AR  
18.    9.    9.    8.    5.    2.3    0.168    0.168    4.04  
PR DEL H MACH AREA R HUB R TIP NB UTIPMAX STR WEIGHT TIN TMAX STAGE I  
1.2550 15.1 0.429 1.849 8.71 12.67 121 1198.8 20998. 50. 844. 844. 1448.

STAGE    4  
WD    WB    WS    WN    WC    CL    RHOB    RHOD    AR  
19.    7.    7.    7.    5.    2.0    0.168    0.168    3.81  
PR DEL H MACH AREA R HUB R TIP NB UTIPMAX STR WEIGHT TIN TMAX STAGE I  
1.2366 15.1 0.418 1.559 9.44 12.67 140 1198.8 17722. 43. 906. 906. 1439.

STAGE    5  
WD    WB    WS    WN    WC    CL    RHOB    RHOD    AR  
18.    5.    5.    6.    4.    1.8    0.168    0.168    3.58  
PR DEL H MACH AREA R HUB R TIP NB UTIPMAX STR WEIGHT TIN TMAX STAGE I  
1.2206 15.1 0.407 1.331 9.98 12.67 158 1198.8 15149. 38. 967. 967. 1382.

STAGE    6  
WD    WB    WS    WN    WC    CL    RHOB    RHOD    AR  
16.    4.    4.    6.    4.    1.6    0.168    0.168    3.35  
PR DEL H MACH AREA R HUB R TIP NB UTIPMAX STR WEIGHT TIN TMAX STAGE I  
1.2068 15.1 0.396 1.150 10.39 12.67 174 1198.8 13100. 34. 1027. 1027. 1324.

STAGE    7  
WD    WB    WS    WN    WC    CL    RHOB    RHOD    AR  
15.    3.    3.    5.    3.    1.5    0.168    0.168    3.12  
PR DEL H MACH AREA R HUB R TIP NB UTIPMAX STR WEIGHT TIN TMAX STAGE I  
1.1947 15.1 0.386 1.004 10.71 12.67 188 1198.8 11443. 31. 1088. 1088. 1260.

STAGE    8

WD	WB	WS	WN	WC	CL	RHOB	RHOD	AR	PR	DEL H	MACH	AREA	R HUB	R TIP	NB UTIPMAX	STR	WEIGHT	TIN	TMAX	STAGE I
14.	3.	3.	5.	3.	1.4	0.168	0.168	2.88												
1.1840	15.1	0.375	0.885	10.96	12.67	200	1198.8	10088.	28.	1147.	1147.									1191.

STAGE 9

WD	WB	WS	WN	WC	CL	RHOB	RHOD	AR	PR	DEL H	MACH	AREA	R HUB	R TIP	NB UTIPMAX	STR	WEIGHT	TIN	TMAX	STAGE I
25.	4.	4.	5.	3.	1.3	0.286	0.286	2.65												
1.1744	15.1	0.364	0.787	11.16	12.67	209	1198.8	15265.	42.	1206.	1206.									2142.

STAGE 10

WD	WB	WS	WN	WC	CL	RHOB	RHOD	AR	PR	DEL H	MACH	AREA	R HUB	R TIP	NB UTIPMAX	STR	WEIGHT	TIN	TMAX	STAGE I
23.	4.	4.	5.	3.	1.3	0.286	0.286	2.42												
1.1659	15.1	0.354	0.704	11.33	12.67	215	1198.8	13673.	39.	1265.	1265.									2013.

STAGE 11

WD	WB	WS	WN	WC	CL	RHOB	RHOD	AR	PR	DEL H	MACH	AREA	R HUB	R TIP	NB UTIPMAX	STR	WEIGHT	TIN	TMAX	STAGE I
22.	4.	4.	5.	3.	1.3	0.286	0.286	2.19												
1.1582	15.1	0.343	0.635	11.47	12.67	217	1198.8	12333.	37.	1324.	1324.									1913.

STAGE 12

WD	WB	WS	WN	WC	CL	RHOB	RHOD	AR	PR	DEL H	MACH	AREA	R HUB	R TIP	NB UTIPMAX	STR	WEIGHT	TIN	TMAX	STAGE I
21.	3.	3.	5.	3.	1.3	0.286	0.286	1.96												
1.1513	15.1	0.332	0.577	11.59	12.67	215	1198.8	11198.	35.	1382.	1382.									1842.

STAGE 13

WD	WB	WS	WN	WC	CL	RHOB	RHOD	AR	PR	DEL H	MACH	AREA	R HUB	R TIP	NB UTIPMAX	STR	WEIGHT	TIN	TMAX	STAGE I
20.	3.	3.	5.	3.	1.3	0.286	0.286	1.73												
1.1450	15.1	0.321	0.527	11.68	12.67	208	1198.8	10230.	34.	1439.	1439.									1796.

STAGE 14

WD	WB	WS	WN	WC	CL	RHOB	RHOD	AR	PR	DEL H	MACH	AREA	R HUB	R TIP	NB UTIPMAX	STR	WEIGHT	TIN	TMAX	STAGE I
19.	3.	3.	5.	3.	1.4	0.286	0.286	1.50												
1.1392	15.1	0.311	0.484	11.77	12.67	197	1198.8	9399.	34.	1496.	1496.									1776.

N STG WEIGHT LENGTH CENGRA INERTIA  
14 581.92 24.74 13.9 22295.5

DUCT  
M NO VEL T TOT P TOT P STAT AREA GAM  
0.311 584. 1553. 64238. 60225. 0.4352 1.3475

PR AD EF PO TO HP  
12.7500 0.8670 64237.9 1553.3 56017.  
HI HO WI CHI  
172.54 383.29 187.86 93.01

\*\*\*\*\* TOTAL COMP WEIGHT IS 581.918

\*\*\*\*\*  
-x x  
x DUCT 17 x  
x x  
\*\*\*\*\*2

MAX CONDITIONS OCCUR AT  
 \*\*\*\*\*

ALT MH  
 PTOT 0. 0.000  
 TTOT 0. 0.000  
 \*\*\*\*\*  
 DUCT , 17  
 RH= 11.90 RT= 12.60 LENG= 6.27  
 AREA= 0.376 RHO=.286  
 CAS WT INC WT WTOT  
 11.3975 10.7609 22.1584

\*\*\*\*\*  
 \* \*  
 \* PBUR 9 \*  
 \* \*  
 \*\*\*\*\*2

MAX CONDITIONS OCCUR AT

\*\*\*\*\*  
 ALT MH VALUE  
 PTOT 0. 0.000 446.1 LB/SQIN  
 TTOT 0. 0.000 1553.3 DEG R  
 CMIN 0. 0.000 9.0 LB/SEC  
 \*\*\*\*\*

BURNER NUMBER 9  
 RIN ROUT LENGTH MACH WSPEC  
 10.177 14.023 21.600 0.053 4.419  
 CAS WT LIN WT NOZ WT INC WT FRAME WTOT  
 43.0 54.2 21.7 31.2 178.6 328.7

\*\*\*\*\*  
 \* \*  
 \* HPT 10 \*  
 \* \*  
 \*\*\*\*\*2

MAX CONDITIONS OCCUR AT

\*\*\*\*\*  
 ALT MH VALUE  
 PTOT 0. 0.000 424.7 LB/SQIN  
 TTOT 0. 0.000 2990.0 DEG R  
 CMOUT 0. 0.000 55.4 LB/SEC  
 \*\*\*\*\*

DUCT  
 M NO VEL T TOT P TOT P STAT AREA GAM  
 0.400 998. 2877. 61154. 55218. 0.4795 1.2912

UTIPMAX STRESS DEN W/AREA TR H/T  
 1060.5 10488.8 0.286 0.294 1.000 0.908

TURBINE 10 MECHANICAL DESIGN  
 H/T H STG LOADING AREA  
 0.908 2.000 0.200 0.479  
 UT RTIP RHUB DEL H RPM MAXRPM TORQ

1060.5 11.2 10.2 224.6 10838.5 10838.5 325775.

STAGE 1  
DISK BLADE VANE HWD CASE AR  
6.9 2.6 13.2 26.1 4.8 1.50  
PR DEL H MACH AREA R HUB R TIP NB MAXUTIP STR WEIGHT LENGTH STAGE I

\*\*\*\* WARNING FOLLOWING STAGE DESIGN LIMIT EXCEEDED \*\*\*\*  
STAGE LOADING IS 0.20 DES LIMIT 150.28  
\*\*STAGE LOADING IS TOO HIGH INCREASE LOADING INPUT 1/TLP=LOADING \*\*  
1.9609 112.3 0.400 0.479 10.18 11.21 154 1060.5 10489. 53.65 2.40 634.

STAGE 2  
DISK BLADE VANE HWD CASE AR  
11.5 7.2 36.2 42.1 8.3 1.50  
PR DEL H MACH AREA R HUB R TIP NB MAXUTIP STR WEIGHT LENGTH STAGE I

\*\*\*\* WARNING FOLLOWING STAGE DESIGN LIMIT EXCEEDED \*\*\*\*  
STAGE LOADING IS 0.20 DES LIMIT 150.28  
\*\*STAGE LOADING IS TOO HIGH INCREASE LOADING INPUT 1/TLP=LOADING \*\*  
2.1890 112.3 0.450 0.796 10.18 11.84 101 1120.1 17422. 105.35 3.88 1356.

N STG LENGTH WEIGHT CENGRA INERTIA  
2 6.28 159.00 5.0 1990.

DUCT  
M NO VEL I TOT P TOT P STAT AREA GAM  
0.500 1065. 2096. 14225. 12113. 1.5459 1.3103

PR TR AD EF PO TO TO.1  
4.2991 1.3553 0.9190 14224.8 2122.4 2096.4  
H IN H OUT AREA FLOW HP  
777.55 552.92 5.23 176.28 56024.

\*\*\*\*\* TOTAL TURB WEIGHT IS 158.997

\*\*\*\*\*  
\* \*  
\* DUCT 11 \*  
\* \*  
\*\*\*\*\*2

MAX CONDITIONS OCCUR AT  
\*\*\*\*\*

ALT MH  
PTOT 0. 0.000  
TTOT 0. 0.000  
\*\*\*\*\*

DUCT , 11  
RH= 10.00 RT= 13.07 LENG= 10.74  
AREA= 1.546 RHO=.286  
CAS WT IHC WT WTOT  
4.6581 3.5643 8.2224

1  
\*\*\*\*\*  
\* \*  
\* LPT 12 \*



X  
\*\*\*\*\*2

# MAX CONDITIONS OCCUR AT

\*\*\*\*\*  
ALT MH VALUE  
PTOT 0. 0.000 98.8 LB/SQIN  
TTOT 0. 0.000 2096.5 DEG R  
CWOUT 0. 0.000 211.0 LB/SEC  
\*\*\*\*\*

## DUCT

M NO VEL T TOT P TOT P STAT AREA GAM  
0.500 1065. 2097. 14229. 12117. 1.5455 1.3103

UTIPMAX STRESS DEN W/AREA TR H/T  
899.1 6967.9 0.286 0.505 1.000 0.916

## TURBINE 12 MECHANICAL DESIGN

H/T H STG LOADING AREA  
0.916 3.000 0.300 1.546  
UT RTIP RHUB DEL H RPM MAXRPM TORQ  
899.1 20.9 19.2 161.5 4920.4 4920.4 542172.

## STAGE 1

DISK BLADE VANE HWD CASE AR  
30.8 13.7 68.5 119.2 11.7 2.00  
PR DEL H MACH AREA R HUB R TIP HB MAXUTIP STR WEIGHT LENGTH STAGE I  
1.5488 53.8 0.500 1.546 19.17 20.94 223 899.1 6968. 243.83 3.10 10705.

## STAGE 2

DISK BLADE VANE HWD CASE AR  
38.3 18.9 94.4 105.2 11.2 3.00  
PR DEL H MACH AREA R HUB R TIP HB MAXUTIP STR WEIGHT LENGTH STAGE I  
1.6212 53.8 0.533 2.178 18.40 20.94 233 899.1 9819. 267.93 2.97 12920.

## STAGE 3

DISK BLADE VANE HWD CASE AR  
45.1 32.8 164.1 103.5 12.8 4.00  
PR DEL H MACH AREA R HUB R TIP HB MAXUTIP STR WEIGHT LENGTH STAGE I  
1.7172 53.8 0.567 3.209 17.07 20.94 203 899.1 14469. 358.29 3.40 16289.

FRAME WT = 318.95

H STG LENGTH WEIGHT CENGRA INERTIA  
3 12.62 1189.01 8.2 39914.

## DUCT

M NO VEL T TOT P TOT P STAT AREA GAM  
0.600 1089. 1525. 3295. 2609. 5.1741 1.3350

PR TR AD EF PO TO TO.1  
4.3186 1.3760 0.9120 3294.8 1523.7 1524.6  
H IN H OUT AREA FLON HP  
544.69 383.23 20.12 185.29 42328.

\*\*\*\*\* TOTAL TURB WEIGHT IS 1189.008

```

*****
*
* DUCT- 13 *
*
*****2

```

```

MAX CONDITIONS OCCUR AT
*****
      ALT      MN
PTOT    0.      0.000
TTOT    0.      0.000
*****

```

```

*****
*
* NOZ  14 *
*
*****2

```

```

MAX CONDITIONS OCCUR AT
*****
      ALT      MN
PTOT    0.      0.000
TTOT    0.      0.000
*****

```

```

NOZZLE  14
WEIGHT= 155.91 LENGTH= 41.876 TR WT= 0.00

```

```

*****
*
* DUCT  4 *
*
*****2

```

```

MAX CONDITIONS OCCUR AT
*****
      ALT      MN
PTOT    0.      0.000
TTOT    0.      0.000
*****

```

```

DUCT , 4
RH= 22.33 RT= 33.92 LENG= 11.59
AREA= 14.224 RHO=.168
      CAS WT      INC WT      WTOT
      7.1743      4.7223      11.8966

```

```

*****
*
* NOZ   5 *
*
*****2

```

```

- MAX CONDITIONS OCCUR AT
-*****
      ALT      MN
PTOT    0.      0.000
TTOT    0.      0.000

```

\*\*\*\*\*  
 NOZZLE 5  
 WEIGHT= 242.87 LENGTH= 67.836 TR WT= 0.00

\*\*\*\*\*  
 \* \*  
 \* SHAF 15 \*  
 \* \*  
 \*\*\*\*\*2

MAX TORQUE CONDITION  
 \*\*\*\*\*  
 TORQUE

10.6  
 \*\*\*\*\*

SHAFT 15  

DO	DI	LENG	DN	WT
6.89	5.86	89.48	0.86	277.62

TOTAL INERTIA OF THIS SPOOL IS 231684.

\*\*\*\*\*  
 \* \*  
 \* SHAF 16 \*  
 \* \*  
 \*\*\*\*\*2

MAX TORQUE CONDITION  
 \*\*\*\*\*  
 TORQUE

9.3  
 \*\*\*\*\*

SHAFT 16  

DO	DI	LENG	DN	WT
7.72	7.29	27.87	2.12	42.22

TOTAL INERTIA OF THIS SPOOL IS 24286.  
 HE DN VALUE OF 2.12 MILLION IS HIGH

\*\*\*\*\*  
 \* \*  
 \* ACCS WT \*  
 \* \*  
 \*\*\*\*\*2

— ACCS WT= 440.164

WEIGHT INPUT DATA IN ENGL UNITS  
WEIGHT OUTPUT DATA IN ENGL UNITS

ENGINE SCALING DATA  
\*\*\*\*\*  
SCALE FACTOR 1.00  
\*\*\*\*\*

COMP NO	WT EST	COMP LEN	ACCU LEN	UPSTREAM RI RO	RADIUS RI RO	DOWNSTREAM RI RO	RADIUS RI RO	NSTAGE
1	0.	0.	0.	0.	0.	0.	0.	0
2	1455.	10.	10.	14.	37.	0.	0.	1
3	0.	0.	10.	19.	34.	0.	0.	0
4	12.	12.	22.	22.	34.	0.	0.	0
5	243.	68.	90.	0.	34.	0.	0.	0
6	314.	4.	14.	20.	23.	0.	0.	1
7	12.	16.	30.	20.	23.	0.	0.	0
8	582.	25.	55.	6.	13.	0.	0.	14
9	329.	22.	83.	10.	14.	0.	0.	0
10	159.	6.	89.	10.	11.	0.	0.	2
11	8.	11.	100.	10.	13.	0.	0.	0
12	1189.	13.	112.	19.	21.	0.	0.	3
13	0.	0.	112.	14.	21.	0.	0.	0
14	156.	42.	154.	0.	21.	0.	0.	0
15	278.	0.	89.	14.	37.	0.	0.	0
16	42.	0.	0.	20.	23.	12.	13.	0
17	22.	6.	61.	12.	13.	0.	0.	10

TOTAL BARE ENGINE WEIGHT= 4800. ACCESSORIES= 440.16 ESTIMATED TOTAL LENGTH= 154. ESTIMATED MAXIMUM RADIUS= 37.  
ESTIMATED CENTER OF GRAVITY= 51.  
ESTIMATED AIRFLOW SCALING EXPONENT IS 1.000



WEIGHT INPUT DATA IN ENGL UNITS  
WEIGHT OUTPUT DATA IN ENGL UNITS

ENGINE SCALING DATA  
\*\*\*\*\*  
SCALE FACTOR 0.80  
\*\*\*\*\*

COMP NO	WT EST	COMP LEN	ACCU LEN	UPSTREAM RADIUS		DOWNSTREAM RADIUS						NSTAGE
				RI	RO	RI	RO	RI	RO	RI	RO	
1	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0
2	1223.	9.	9.	12.	33.	0.	0.	17.	31.	0.	0.	1
3	0.	0.	9.	17.	31.	0.	0.	17.	21.	21.	31.	0
4	9.	10.	20.	20.	31.	0.	0.	20.	31.	0.	0.	0
5	197.	61.	81.	0.	31.	0.	0.	0.	29.	0.	0.	0
6	223.	4.	13.	18.	21.	0.	0.	18.	21.	0.	0.	1
7	8.	14.	27.	18.	21.	0.	0.	18.	21.	0.	0.	0
8	419.	22.	49.	5.	11.	0.	0.	11.	11.	0.	0.	16
9	268.	22.	76.	9.	13.	0.	0.	9.	13.	0.	0.	0
10	115.	6.	82.	9.	10.	0.	0.	9.	12.	0.	0.	2
11	6.	10.	92.	9.	12.	0.	0.	9.	12.	0.	0.	0
12	914.	11.	103.	17.	19.	0.	0.	13.	19.	0.	0.	3
13	0.	0.	103.	13.	19.	0.	0.	13.	19.	0.	0.	0
14	125.	37.	140.	0.	19.	0.	0.	0.	17.	0.	0.	0
15	237.	0.	82.	12.	33.	0.	0.	9.	12.	0.	0.	0
16	38.	0.	0.	18.	21.	11.	11.	0.	0.	0.	0.	0
17	16.	6.	55.	11.	11.	0.	0.	11.	11.	0.	0.	10

TOTAL BARE ENGINE WEIGHT= 3799. ACCESSORIES= 347.68 ESTIMATED TOTAL LENGTH= 140. ESTIMATED MAXIMUM RADIUS= 33.  
ESTIMATED CENTER OF GRAVITY= 46.  
ESTIMATED AIRFLOW SCALING EXPONENT IS 1.049

WEIGHT INPUT DATA IN ENGL UNITS  
WEIGHT OUTPUT DATA IN ENGL UNITS

ENGINE SCALING DATA  
\*\*\*\*\*  
SCALE FACTOR 1.20  
\*\*\*\*\*

COMP NO	WT EST	COMP LEN	ACCU LEN	UPSTREAM RADIUS		DOWNSTREAM RADIUS		NSTAGE
			RI	RO	RI	RO		
1	0.	0.	0.	0.	0.	0.	0.	0
2	1707.	11.	11.	15.	41.	0.	0.	1
3	0.	0.	11.	21.	38.	0.	0.	0
4	16.	13.	24.	24.	37.	0.	0.	0
5	288.	74.	98.	0.	37.	0.	0.	0
6	415.	4.	16.	22.	26.	0.	0.	1
7	16.	17.	33.	22.	25.	0.	0.	0
8	766.	27.	60.	7.	14.	0.	0.	14
9	389.	22.	89.	11.	15.	0.	0.	0
10	208.	7.	96.	11.	12.	0.	0.	2
11	11.	12.	107.	11.	14.	0.	0.	0
12	1489.	14.	121.	21.	23.	0.	0.	3
13	0.	0.	121.	16.	23.	0.	0.	0
14	187.	46.	167.	0.	23.	0.	0.	0
15	316.	0.	96.	15.	41.	0.	0.	0
16	46.	0.	0.	22.	25.	13.	14.	0
17	29.	7.	67.	13.	14.	0.	0.	10

TOTAL BARE ENGINE WEIGHT= 5882. ACCESSORIES= 540.64 ESTIMATED TOTAL LENGTH= 167. ESTIMATED MAXIMUM RADIUS= 41.  
ESTIMATED CENTER OF GRAVITY= 56.  
ESTIMATED AIRFLOW SCALING EXPONENT IS 1.114

UPDATED INPUT DATA TO REFLECT CALCULATED INPUT  
COMPONENT

NO.	TYPE	DATINP1	DATINP2	DATINP3	DATINP4	DATINP5	DATINP6	DATINP7	DATINP8	DATINP9
1	INLET	0.10000D 04	0.00000	0.14696D 02	0.00000	0.00000	0.10000D 01	0.00000	0.00000	0.10000D 04
2	COMPRESR	0.18000D 01	0.00000	0.39250D 04	0.37610D 04	0.10227D 04	0.37620D 04	0.95772D 00	0.37630D 04	0.39127D 00
3	SPLITTER	0.43230D 01	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
4	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
5	NOZZLE	0.15353D 04	0.10000D 01	0.00000	0.00000	0.99000D 00	0.00000	0.00000	0.00000	0.10000D 01
6	COMPRESR	0.13000D 01	0.00000	0.35737D 04	0.37610D 04	0.12437D 03	0.37620D 04	0.10023D 01	0.37630D 04	0.18266D 00
7	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
8	COMPRESR	0.13000D 01	0.16200D 00	0.51680D 04	0.37070D 04	0.92177D 02	0.37080D 04	0.90596D 00	0.37090D 04	0.18516D 01
9	DUCT B	0.48000D 01	0.00000	0.00000	0.29900D 04	0.10000D 01	0.18400D 05	0.00000	0.00000	0.00000
10	TURBINE	0.40000D 01	0.79000D 00	0.43996D 00	0.38010D 04	0.68035D 00	0.38020D 04	0.10336D 01	0.10993D 01	0.62500D 00
11	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
12	TURBINE	0.25000D 01	0.21000D 00	0.37940D 00	0.38030D 04	0.86961D 00	0.38040D 04	0.99700D 00	0.22118D 01	0.00000
13	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
14	NOZZLE	0.64017D 03	0.10000D 01	0.00000	0.00000	0.99000D 00	0.00000	0.00000	0.00000	0.10000D 01
15	SHAFT	0.40000D 04	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01
16	SHAFT	0.60000D 04	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01	0.10000D 01
17	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000

## CASE IDENTIFICATION      SAMPLE CASE FOR PREPWATE

## STATION PROPERTY OUTPUT DATA

FLOW STATION	WEIGHT FLOW STATP1	TOTAL PRESSURE STATP2	TOTAL TEMPERATURE STATP3	FUEL/AIR RATIO STATP4	REFERRED FLOW STATP5	MACH NUMBER STATP6	STATIC PRESSURE STATP7	INTERFACE CORRECTED FLOW ERROR STATP8
1	0.10000D 04	0.14696D 02	0.53867D 03	0.00000	0.10191D 04	0.00000	0.00000	0.00000
2	0.10000D 04	0.14696D 02	0.53867D 03	0.00000	0.10191D 04	0.00000	0.00000	0.00000
3	0.10000D 04	0.25483D 02	0.64979D 03	0.00000	0.64548D 03	0.00000	0.00000	0.00000
4	0.81214D 03	0.25483D 02	0.64979D 03	0.00000	0.52422D 03	0.00000	0.00000	0.00000
5	0.81214D 03	0.25483D 02	0.64979D 03	0.00000	0.52422D 03	0.92227D 00	0.14696D 02	0.00000
6	0.81214D 03	0.25483D 02	0.64979D 03	0.00000	0.52422D 03	0.91304D 00	0.14696D 02	0.00000
7	0.18786D 03	0.25483D 02	0.64979D 03	0.00000	0.12126D 03	0.00000	0.00000	0.00000
8	0.18786D 03	0.34988D 02	0.72056D 03	0.00000	0.93005D 02	0.00000	0.00000	0.00000
9	0.18786D 03	0.34988D 02	0.72056D 03	0.00000	0.93005D 02	0.00000	0.00000	0.00000
10	0.15743D 03	0.44610D 03	0.15533D 04	0.00000	0.89748D 01	0.00000	0.00000	0.00000
11	0.16125D 03	0.42468D 03	0.29900D 04	0.24256D-01	0.13397D 02	0.00000	0.00000	0.00000
12	0.18529D 03	0.98812D 02	0.20965D 04	0.21043D-01	0.55404D 02	0.00000	0.00000	0.00000
13	0.18529D 03	0.98812D 02	0.20965D 04	0.21043D-01	0.55404D 02	0.00000	0.00000	0.00000
14	0.19168D 03	0.22886D 02	0.15247D 04	0.20327D-01	0.21104D 03	0.00000	0.00000	0.00000
15	0.19168D 03	0.22886D 02	0.15247D 04	0.20327D-01	0.21104D 03	0.83092D 00	0.14696D 02	0.00000
16	0.19168D 03	0.22886D 02	0.15247D 04	0.20327D-01	0.21104D 03	0.82261D 00	0.14696D 02	0.00000
17	0.30434D 02	0.44610D 03	0.15533D 04	0.00000	0.00000	0.00000	0.00000	0.00000
19	0.15743D 03	0.44610D 03	0.15533D 04	0.00000	0.89748D 01	0.00000	0.00000	0.00000

## COMPONENT OUTPUT DATA

COMPONENT NO. TYPE	DATOUT1	DATOUT2	DATOUT3	DATOUT4	DATOUT5	DATOUT6	DATOUT7	DATOUT8	DATOUT9
1 INLET	0.00000	0.00000	0.00000	0.10000D 01	0.10000D 01	0.00000	0.10000D 01	0.10386D 01	0.10000D-04
2 COMPRESR	-0.37786D 05	0.40000D 04	0.00000	0.18000D 01	0.81382D 01	0.10000D 01	0.10227D 04	0.82400D 00	0.17340D 01
3 SPLITTER	0.43230D 01	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
4 DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
5 NOZZLE	0.26647D 05	0.10556D 04	0.17340D 01	0.15353D 04	0.15353D 04	0.10000D 01	0.99000D 00	0.18930D 01	0.17340D 01
6 COMPRESR	-0.45423D 04	0.40000D 04	0.00000	0.13000D 01	0.16143D 01	0.10000D 01	0.12437D 03	0.86700D 00	0.13730D 01
7 DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
8 COMPRESR	-0.56024D 05	0.60000D 04	0.00000	0.13000D 01	0.32396D 02	0.98500D 00	0.92177D 02	0.86700D 00	0.12750D 02
9 DUCT B	0.00000	0.48000D-01	0.00000	0.24256D-01	0.00000	0.13747D 05	0.00000	0.10000D 01	0.29900D 04
10 TURBINE	0.56024D 05	0.60000D 04	0.10000D 01	0.40000D 01	0.43996D 00	0.56800D 04	0.68035D 00	0.91900D 00	0.42979D 01
11 DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
12 TURBINE	0.42328D 05	0.40000D 04	0.10000D 01	0.25000D 01	0.37940D 00	0.52440D 04	0.86961D 00	0.91200D 00	0.43176D 01
13 DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
14 NOZZLE	0.87342D 04	0.14660D 04	0.15573D 01	0.64017D 03	0.64017D 03	0.10000D 01	0.99000D 00	0.18616D 01	0.15573D 01
15 SHAFT	0.00000	0.40000D 04	0.40000D 04	0.40000D 04	0.40000D 04	0.00000	0.00000	0.00000	0.00000
16 SHAFT	0.00000	0.60000D 04	0.60000D 04	0.60000D 04	0.00000	0.00000	0.00000	0.00000	0.00000
17 DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000

MACH= 0.0000    ALTITUDE= 0.    RECOVERY= 1.0000    0 ITERATIONS    2 PASSES

AIRFLOW (LB/SEC)	1000.00	GROSS THRUST	35380.75	FUEL FLOW (LB/HR)	13747.27
NET THRUST	35380.75	TSFC	0.3886	NET THRUST/AIRFLOW	35.3808
TOTAL INLET DRAG	0.00	TOTAL BRAKE SHAFT HP	0.00	BOAT TAIL DRAG	0.00
INSTALLED THRUST	35380.75	INSTALLED TSFC	0.3886	SPILLAGE + LIP DRAG	0.00



## APPENDIX D

## Source Code Listing for PREPWATE

```

IMPLICIT REAL*8 (A-H,O-Z)                                0000100
LOGICAL IPLT,PLOT,ISII,ISIO                                0000200
DIMENSION ILENG(30), IREST(30), IALL(60), ITYPE(60), ITYPEI(60), I 0000300
IWORD(11), DEFAULT(17,18), DESVAL(17,60), RMIN(17,18), RMAX(17,18), 0000400
2IWMEC(7,60), RMAN(7), RMIX(7)                             0000500
DATA IWORD/4HINLT,4HDUCT,4HWINJ,4HCOMP,4HTURB,4HHTEX,4HSPLT,4HMIXR 0000600
1,4HNOZZ,4HLOAD,4HSHFT/                                     0000700
INTEGER AA,BB,CC,DD,EE,FF,GG,HH,OO,PP,QQ,RR,SS,TT,PBURX,DBURX,AUGX 0000800
1,DUCTX,FANX,FOX,FIX,RSFOX,RSFIX,HPCX,HPTX,HTXX,FMIX,SHAFX      0000900
DATA AA,BB,CC,DD,EE,FF,GG,HH,II,JJ,KK,LL,MM,NN,OO,PP,QQ,RR,SS,TT/1 0001000
1HA,1HB,1HC,1HD,1HE,1HF,1HG,1HH,1HI,1HJ,1HK,1HL,1HM,1HN,1HO,1HP,1HQ 0001100
2,1HR,1HS,1HT/                                              0001200
DATA PBURX,DBURX,AUGX,DUCTX/4HPBUR,4HDBUR,4HAUG,4HDUCT/        0001300
DATA FMIX,MIX,HOZX,SHAFX/4HFMIX,4HMIX,4HNOZ,4HSHAF/            0001400
DATA FANX,FOX,FIX,RSFOX,RSFIX,LPCX,HPCX/4HFAN,4HFO,4HFI,4HRSF 0001500
10,4HRSFI,4HLPC,4HHPC,/,HPTX,LPTX,HTXX/4HHPT,4HLPT,4HHTEX/    0001600
DATA DEFAULT/100,.,.015,15*0,150,.,.015,15*0,300,.,.015,15*0,.,0.4,1., 0001700
10,.,.1,13*0,.,.55,1.7,.,.45,1.5,4,.,.3,.,.45,0,.,.1,0,.,.2,1,2*0,.,1.8, 0001800
2.055,.,.55,1.7,.,.45,1.5,4,.,.3,.,.45,0,.,.1,0,.,.2,1,2*0,.,1.8,.,.055,.,.55 0001900
3.1,7,.,.45,1.5,4,.,.3,.,.45,0,.,.1,0,.,.2,1,2*0,.,1.8,.,.055,.,.55,1.7,.,.45 0002000
4.1,5,4,.,.3,.,.45,0,.,.1,0,.,.2,1,2*0,.,1.8,.,.055,.,.55,1.7,.,.45,1.5,4,., 0002100
53,.,.45,0,.,.1,0,.,.2,1,2*0,.,1.8,.,.055,.,.50,1.5,40,1.5,4,.,.3,.,.45,0 0002200
6,0,.,.1,0,.,.2,1,2*0,.,1.2,120,40,1.4,70,1.5,3,1.5,30,0,.,.1, 0002300
7,0,.,.2,1,2*0,.,1.2,120,3,28,1.5,1.5,1.5,45,125000,2,1,2*0,., 0002400
80,155,3*0,.,.1,0,0.155,.,.45,28,1.5,2,4,.,.55,125000,2,1,2*0,.,0.19 0002500
95,3*0,.,.1,0,0.195,.,.3,4*0,.,.45,11*0,5000,.,.5,5,14*0,.,.1,8,.,.15*0,.,1 0002600
0,.,.16*0,.,.50000,.,.286,15*0,./ 0002700
DATA RMIN/100,.,.01,15*0,.,.150,.,.01,15*0,.,.200,.,.01,15*0,.,.4,16*0,.,.5 0002800
1,1,5,4,1,3,2,.,.45,0,.,.1,0,.,.1,4*0,.,.5,1,5,4,1,3,2,.,.45 0002900
2,0,.,.1,0,.,.1,4*0,.,.5,1,5,4,1,3,2,.,.45,0,.,.1,0,.,.1,4 0003000
3*0,.,.5,1,5,4,1,3,2,.,.45,0,.,.1,0,.,.1,4*0,.,.5,1,5,4,1,3, 0003100
42,.,.45,0,.,.1,0,.,.1,4*0,.,.45,1,5,4,1,3,2,.,.45,0,.,.1,0, 0003200
50,1,4*0,.,.4,1,4,6,1,2,1,20,0,.,.1,0,.,.1,4*0,.,.3,2,1, 0003300
61,1,.,.45,100000,.,.0,1,8*0,.,.4,2,1,2,4,.,.55,100000,.,.0,1,8*0, 0003400
7,17*0,.,.0,3,3,14*0,.,.1,7,15*0,.,.1,16*0,.,.40000,.,.28,15*0,./ 0003500
DATA RMAX/150,.,.02,15*0,.,.200,.,.02,15*0,.,.300,.,.02,15*0,.,.5,16*0,.,.6 0003600
1,1,8,5,1,5,5,3,.,.55,0,.,.1,0,.,.1,4*0,.,.6,1,8,5,1,5,5,3,., 0003700
255,0,.,.1,0,.,.1,4*0,.,.6,1,8,5,1,5,5,3,.,.55,0,.,.1,0,.,.1 0003800
3,4*0,.,.6,1,8,5,1,5,5,3,.,.55,0,.,.1,0,.,.1,4*0,.,.6,1,8,5,1, 0003900
45,5,3,.,.55,0,.,.1,0,.,.1,4*0,.,.6,1,8,5,1,5,5,3,.,.55,0,.,.1 0004000
5,0,.,.1,4*0,.,.5,1,7,8,1,5,5,2,.,.30,0,.,.1,0,.,.1,4*0,.,.4, 0004100
6,3,1,5,2,2,.,.5,150000,.,.0,1,8*0,.,.5,3,1,5,3,6,.,.6,150000,.,.0,1 0004200
7,8*0,.,.17*0,.,.0,5,5,14*0,.,.2,9,15*0,.,.2,16*0,.,.50000,.,.31,85,14 0004300
8*0,./ 0004400
DATA RMAN,RMIX/14*0,./,IPLT,PLOT,ISII,ISIO/.TRUE,.,.TRUE,.,.FALSE,.,.F 0004500
1ALSE,./,IWT,IOUTCD/2,2/ 0004600
WRITE (35,970) 0004700
READ (30,980) IPLT,PLOT,ISII,ISIO 0004800
WRITE (6,990) IPLT,PLOT,ISII,ISIO 0004900
WRITE (35,1000) 0005000
READ (30,1010) IWT,IOUTCD 0005100
WRITE (6,1020) IWT,IOUTCD 0005200
WRITE (35,1210) 0005300
READ (30,1120) IDISK 0005400
IF (IDISK.EQ.88) DISKNI=0. 0005500
IF (IDISK.EQ.66) DISKNI=1. 0005600
WRITE (6,1220) DISKNI 0005700
DO 10 I=1,30 0005800
IWMEC(1,I)=0 0005900
IWMEC(1,I+30)=0 0006000

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10	IREST(I)=0	0006100
	I LENG(I)=0	0006200
	WRITE (35,1030)	0006300
	READ (30,1040) I LENG	0006400
	WRITE (35,1050)	0006500
	READ (30,1040) I REST	0006600
	DO 20 I=1,30	0006700
	IF (I LENG(I).EQ.0) GO TO 30	0006800
20	ILMAX=I	0006900
30	WRITE (6,1060) (I LENG(I),I=1,ILMAX)	0007000
	DO 40 I=1,30	0007100
	IF (IREST(I).EQ.0) GO TO 50	0007200
40	IRMAX=I	0007300
50	IBEG=ILMAX+1	0007400
	IEND=ILMAX+IRMAX	0007500
	DO 60 I=1,60	0007600
60	IALL(I)=0	0007700
	DO 70 I=1,ILMAX	0007800
70	IALL(I)=I LENG(I)	0007900
	DO 80 I=IBEG,IEND	0008000
	IRST=I-IBEG+1	0008100
80	IALL(I)=IREST(IRST)	0008200
C	IALL CONTAINS ALL THE COMPONENT NUMBERS FOR WHICH WE NEED INPUTS	0008300
C	NOW IDENTIFY COMPONENT TYPE PER HNEP GENERIC TYPE	0008400
	WRITE (35,1070)	0008500
	DO 90 I=1,IEND	0008600
	WRITE (35,1080) IALL(I)	0008700
	READ (30,1090) I TYPE(I)	0008800
90	CONTINUE	0008900
	DO 110 I=1,IEND	0009000
	DO 100 J=1,11	0009100
	IF (I TYPE(I).NE.IWORD(J)) GO TO 100	0009200
	I TYPE(I)=J	0009300
	GO TO 110	0009400
100	CONTINUE	0009500
110	CONTINUE	0009600
C	I TYPE(I) HAS COMPONENT TYPE BY NUMERICAL TYPE	0009700
	WRITE (35,1100)	0009800
	JCX=0	0009900
120	JCX=JCX+1	0010000
	JCXX=IALL(JCX)	0010100
	IF (JCX.GT.IEND) GO TO 570	0010200
	IGO=I TYPE(I)(JCX)	0010300
	GO TO (130,140,240,250,330,390,430,440,480,560,520), IGO	0010400
C	INLET YOU DUMMY	0010500
130	GO TO 120	0010600
C	DUCT	0010700
140	WRITE (35,1110) IALL(JCX)	0010800
	READ (30,1120) IJK	0010900
	IF (IJK.EQ.AA) IWMIEC(1,JCXX)=PBURX	0011000
	IF (IJK.EQ.AA) KT=1	0011100
	IF (IJK.EQ.BB) IWMIEC(1,JCXX)=DBURX	0011200
	IF (IJK.EQ.BB) KT=2	0011300
	IF (IJK.EQ.CC) IWMIEC(1,JCXX)=AUGX	0011400
	IF (IJK.EQ.CC) KT=3	0011500
	IF (IJK.EQ.DD) IWMIEC(1,JCXX)=DUCTX	0011600
	IF (IJK.EQ.DD) KT=4	0011700
	IF (KT.EQ.4) GO TO 150	0011800
	WRITE (35,1130)	0011900
	READ (30,1120) IJK	0012000

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      IWMEC(2,JCXX)=8
      IF (IJK.EQ.AA) IWMEC(2,JCXX)=1
      GO TO 160
150  WRITE (35,1140)
      READ (30,1120) IJK
      IF (IJK.EQ.AA) IWMEC(2,JCXX)=1
      IF (IJK.EQ.BB) IWMEC(2,JCXX)=2
      IF (IJK.EQ.CC) IWMEC(2,JCXX)=3
      IF (IJK.EQ.DD) IWMEC(2,JCXX)=4
      IF (IJK.EQ.EE) IWMEC(2,JCXX)=5
C    NOW FILL IN THE DEFAULT VALUES FOR THIS COMPONENT
160  DO 170 I=1,17
170  DESVAL(I,JCXX)=DEFAULT(I,KT)
      KX=JCXX
180  IF (KT.GT.1) GO TO 190
      WRITE (35,1150) AA,DESVAL(1,KX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2,
1KX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,KX),RMIN(3,KT),RMAX(3,KT),DD
2,DESVAL(4,KX),RMIN(4,KT),RMAX(4,KT),EE,DESVAL(5,KX),RMIN(5,KT),RMA
3X(5,KT)
      GO TO 220
190  IF (KT.GT.2) GO TO 200
      WRITE (35,1160) AA,DESVAL(1,KX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2,
1KX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,KX),RMIN(3,KT),RMAX(3,KT),DD
2,DESVAL(4,KX),RMIN(4,KT),RMAX(4,KT),EE,DESVAL(5,KX),RMIN(5,KT),RMA
3X(5,KT)
      GO TO 220
200  IF (KT.GT.3) GO TO 210
      WRITE (35,1170) AA,DESVAL(1,KX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2,
1KX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,KX),RMIN(3,KT),RMAX(3,KT),DD
2,DESVAL(4,KX),RMIN(4,KT),RMAX(4,KT),EE,DESVAL(5,KX),RMIN(5,KT),RMA
3X(5,KT)
      GO TO 220
210  CONTINUE
      WRITE (35,1200) AA,DESVAL(1,KX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2,
1KX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,KX),RMIN(3,KT),RMAX(3,KT),DD
2,DESVAL(4,KX),RMIN(4,KT),RMAX(4,KT)
220  WRITE (35,960) QQ,RR,TT
230  CALL REDDEM (IND,DESVAL,JCXX)
      IF (IND.NE.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 230
      IF (IND.EQ.QQ) GO TO 570
      IF (IND.EQ.TT) GO TO 120
      IF (IND.EQ.RR) GO TO 180
C    NO WEIGHT OR DIMENSIONS FOR WATER INJECTORS
240  CONTINUE
      GO TO 120
250  WRITE (35,950) IALL(JCX)
      READ (30,1120) IJK
      IF (IJK.EQ.AA) IWMEC(1,JCXX)=FANX
      IF (IJK.EQ.AA) KT=5
      IF (IJK.EQ.BB) IWMEC(1,JCXX)=FOX
      IF (IJK.EQ.BB) KT=6
      IF (IJK.EQ.CC) IWMEC(1,JCXX)=FIX
      IF (IJK.EQ.CC) KT=7
      IF (IJK.EQ.DD) IWMEC(1,JCXX)=RSFOX
      IF (IJK.EQ.DD) KT=8
      IF (IJK.EQ.EE) IWMEC(1,JCXX)=RSFIX
      IF (IJK.EQ.EE) KT=9
      IF (IJK.EQ.FF) IWMEC(1,JCXX)=LPCX
      IF (IJK.EQ.FF) KT=10
      IF (IJK.EQ.GG) IWMEC(1,JCXX)=HPCX

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IF (IJK.EQ.GG) KT=11
WRITE (35,850)
READ (30,1120) IJK
IF (IJK.EQ.AA) IWMEC(2,JCXX)=1
IF (IJK.EQ.BB) IWMEC(2,JCXX)=0
IF (IJK.EQ.CC) IWMEC(2,JCXX)=2
WRITE (35,860)
READ (30,1120) IJK
IF (IJK.EQ.AA) IWMEC(3,JCXX)=0
IF (IJK.EQ.BB) IWMEC(3,JCXX)=1
IF (IJK.EQ.CC) IWMEC(3,JCXX)=2
IF (IJK.EQ.DD) IWMEC(3,JCXX)=4
WRITE (35,870)
READ (30,1120) IJK
IF (IJK.EQ.AA) IWMEC(4,JCXX)=0
IF (IJK.EQ.BB) IWMEC(4,JCXX)=1
IF (IJK.EQ.CC) IWMEC(4,JCXX)=2
IF (IJK.EQ.DD) IWMEC(4,JCXX)=4
IF (KT.LT.6.OR.KT.GT.9) GO TO 260
WRITE (35,880)
READ (30,890) IWMEC(5,JCXX)
260 WRITE (35,900)
READ (30,890) IWMEC(6,JCXX)
WRITE (35,910)
READ (30,890) IWMEC(7,JCXX)
C NOW FILL IN THE DEFAULT VALUES FOR THIS COMPONENT
DO 270 I=1,17
270 DESVAL(I,JCXX)=DEFAULT(I,KT)
IF (IWMEC(2,JCXX).NE.2) GO TO 290
DESVAL(4,JCXX)=DESVAL(10,JCXX)
DESVAL(5,JCXX)=DESVAL(7,JCXX)
DESVAL(6,JCXX)=1.
DO 280 IJJ=7,17
280 DESVAL(IJJ,JCXX)=0.
GO TO 310
C NOW WE UPDATE COMPRESSOR DESIGN VALUES
290 WRITE (35,930) AA,DESVAL(1,JCXX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2
1,JCXX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,JCXX),RMIN(3,KT),RMAX(3,
21),DD,DESVAL(4,JCXX),RMIN(4,KT),RMAX(4,KT),EE,DESVAL(5,JCXX),RMIN(
35,KT),RMAX(5,KT),FF,DESVAL(6,JCXX),RMIN(6,KT),RMAX(6,KT),GG,DESVAL
4(7,JCXX),RMIN(7,KT),RMAX(7,KT),HH,DESVAL(8,JCXX),RMIN(8,KT),RMAX(8
5,KT),II,DESVAL(9,JCXX),RMIN(9,KT),RMAX(9,KT),JJ,DESVAL(10,JCXX),RM
6IN(10,KT),RMAX(10,KT),KK,DESVAL(11,JCXX),RMIN(11,KT),RMAX(11,KT),L
71,DESVAL(12,JCXX),RMIN(12,KT),RMAX(12,KT),MM,DESVAL(13,JCXX),RMIN(
813,KT),RMAX(13,KT),NN,DESVAL(14,JCXX),RMIN(14,KT),RMAX(14,KT),OO,D
9ESVAL(15,JCXX),RMIN(15,KT),RMAX(15,KT),PP,DESVAL(16,JCXX),RMIN(16,
6KT),RMAX(16,KT),SS,DESVAL(17,JCXX),RMIN(17,KT),RMAX(17,KT)
WRITE (35,960) QQ,RR,TT
300 CALL REDDEM (IND,DESVAL,JCXX)
IF (IND.NE.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 300
IF (IND.EQ.QQ) GO TO 570
IF (IND.EQ.TT) GO TO 120
IF (IND.EQ.RR) GO TO 290
310 WRITE (35,940) AA,DESVAL(1,JCXX),RMAN(1),RMIX(1),BB,DESVAL(2,JCXX)
1,RMAN(2),RMIX(2),CC,DESVAL(3,JCXX),RMAN(3),RMIX(3),DD,DESVAL(4,JCX
2X),RMAN(4),RMIX(4),EE,DESVAL(5,JCXX),RMAN(5),RMIX(5),FF,DESVAL(6,J
3CXX),RMAN(6),RMIX(6),GG,DESVAL(7,JCXX),RMAN(7),RMIX(7)
WRITE (35,960) QQ,RR,TT
320 CALL REDDEM (IND,DESVAL,JCXX)
IF (IND.NE.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 320

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0021100
0021200
0021300
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0021500
0021600
0021700
0021800
0021900
0022000
0022100
0022200
0022300
0022400
0022500
0022600
0022700
0022800
0022900
0023000
0023100
0023200
0023300
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0023500
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0023700
0023800
0023900
0024000

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      IF (IND.EQ.QQ) GO TO 570
      IF (IND.EQ.TT) GO TO 120
      IF (IND.EQ.RR) GO TO 310
C     COME HERE FOR TURBINES
330  WRITE (35,770) IALL(JCX)
      READ (30,1120) ITURB
      IF (ITURB.EQ.HH) IWMEC(1,JCXX)=HPTX
      IF (ITURB.EQ.LL) IWMEC(1,JCXX)=LPTX
      WRITE (35,780)
      READ (30,1120) IJK
      IF (IJK.EQ.AA) IWMEC(2,JCXX)=0
      IF (IJK.EQ.BB) IWMEC(2,JCXX)=1
      WRITE (35,790)
      READ (30,1040) IWMEC(3,JCXX)
      WRITE (35,800)
      READ (30,820) IWMEC(4,JCXX)
      WRITE (35,830)
      READ (30,1040) IWMEC(5,JCXX)
      WRITE (35,840)
      READ (30,1120) ITURB
      IF (ITURB.EQ.RR) IWMEC(6,JCXX)=2
      IF (ITURB.EQ.AA) IWMEC(6,JCXX)=0
      KT=12
      IF (IWMEC(1,JCXX).EQ.LPTX) KT=13
      IF (IWMEC(6,JCXX).EQ.2) KT=14
      DO 340 I=1,17
340  DESVAL(I,JCXX)=DEFAULT(I,KT)
      IF (KT.EQ.14) GO TO 370
350  WRITE (35,730) AA,DESVAL(1,JCXX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2
1,JCXX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,JCXX),RMIN(3,KT),RMAX(3,KT),
2,DESVAL(4,JCXX),RMIN(4,KT),RMAX(4,KT),EE,DESVAL(5,JCXX),RMIN(
35,KT),RMAX(5,KT),FF,DESVAL(6,JCXX),RMIN(6,KT),RMAX(6,KT),GG,DESVAL
4(7,JCXX),RMIN(7,KT),RMAX(7,KT),HH,DESVAL(8,JCXX),RMIN(8,KT),RMAX(8
5,KT),II,DESVAL(9,JCXX),RMIN(9,KT),RMAX(9,KT),JJ,DESVAL(10,JCXX),RM
6IN(10,KT),RMAX(10,KT),KK,DESVAL(11,JCXX),RMIN(11,KT),RMAX(11,KT),L
7L,DESVAL(12,JCXX),RMIN(12,KT),RMAX(12,KT),PP,DESVAL(16,JCXX),RMIN(
816,KT),RMAX(16,KT),SS,DESVAL(17,JCXX),RMIN(17,KT),RMAX(17,KT)
      WRITE (35,960) QQ,RR,TT
360  CALL REDOEM (IND,DESVAL,JCXX)
      IF (IND.NE.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 360
      IF (IND.EQ.QQ) GO TO 570
      IF (IND.EQ.TT) GO TO 120
      IF (IND.EQ.RR) GO TO 350
370  WRITE (35,740) AA,DESVAL(1,JCXX),RMIN(1,KT),RMAX(1,KT),FF,DESVAL(6
1,JCXX),RMIN(6,KT),RMAX(6,KT)
      WRITE (35,960) QQ,RR,TT
380  CALL REDOEM (IND,DESVAL,JCXX)
      IF (IND.NE.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 380
      IF (IND.EQ.QQ) GO TO 570
      IF (IND.EQ.TT) GO TO 120
      IF (IND.EQ.RR) GO TO 370
C     COME HERE IF HEAT EXCHANGER
390  WRITE (35,750) IALL(JCX)
      IWMEC(1,JCXX)=HTXX
      READ (30,1120) IHX
      IF (IHX.EQ.FF) IWMEC(2,JCXX)=1
      IF (IHX.EQ.RR) IWMEC(2,JCXX)=2
      WRITE (35,760)
      READ (30,1120) IDIR
      IF (IDIR.EQ.PP) IWMEC(3,JCXX)=1

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0024100
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0025000
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0029000
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0029900
0030000

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      IF (IDIR.EQ.CC) INMEC(3,JCXX)=2
      KT=15
      DO 400 I=1,17
      DESVAL(I,JCXX)=DEFAULT(I,KT)
400  WRITE (35,920) AA,DESVAL(1,JCXX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2
410  1,JCXX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,JCXX),RMIN(3,KT),RMAX(3,
      2T),DD,DESVAL(4,JCXX),RMIN(4,KT),RMAX(4,KT)
      WRITE (35,960) QQ,RR,TT
420  CALL REDDEM (IND,DESVAL,JCXX)
      IF (IND.NE.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 420
      IF (IND.EQ.QQ) GO TO 570
      IF (IND.EQ.TT) GO TO 120
      IF (IND.EQ.RR) GO TO 410
      C
      COME HERE IF SPLITTER
430  WRITE (35,680) IALL(JCX)
      READ (30,1010) INMEC(2,JCXX)
      INMEC(1,JCXX)=INWORD(7)
      IF (JCXX.HE.ILENG(1)) GO TO 120
      C
      FIRST COMPONENT
      WRITE (35,690)
      READ (30,700) DESVAL(1,JCXX),DESVAL(2,JCXX)
      GO TO 120
      C
      COME HERE IF MIXER
440  WRITE (35,710) IALL(JCX)
      READ (30,1120) IMIXR
      IF (IMIXR.EQ.FF) INMEC(1,JCXX)=FMIX
      IF (IMIXR.EQ.CC) INMEC(1,JCXX)=MIX
      WRITE (35,720)
      READ (30,1010) INMEC(2,JCXX)
      KT=16
      DO 450 I=1,17
      DESVAL(I,JCXX)=DEFAULT(I,KT)
450  WRITE (35,600) AA,DESVAL(1,JCXX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2
460  1,JCXX),RMIN(2,KT),RMAX(2,KT)
      WRITE (35,960) QQ,RR,TT
470  CALL REDDEM (IND,DESVAL,JCXX)
      IF (IND.NE.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 470
      IF (IND.EQ.QQ) GO TO 570
      IF (IND.EQ.TT) GO TO 120
      IF (IND.EQ.RR) GO TO 460
      C
      COME HERE IF NOZZLE
480  WRITE (35,610) IALL(JCX)
      INMEC(1,JCXX)=NOZX
      READ (30,1010) INMEC(2,JCXX)
      WRITE (35,810)
      READ (30,820) INMEC(3,JCXX)
      WRITE (35,620)
      READ (30,1010) INMEC(4,JCXX)
      KT=17
      DO 490 I=1,17
      DESVAL(I,JCXX)=DEFAULT(I,KT)
490  WRITE (35,630) AA,DESVAL(1,JCXX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2
500  1,JCXX),RMIN(2,KT),RMAX(2,KT)
      WRITE (35,960) QQ,RR,TT
510  CALL REDDEM (IND,DESVAL,JCXX)
      IF (IND.NE.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 510
      IF (IND.EQ.QQ) GO TO 570
      IF (IND.EQ.TT) GO TO 120
      IF (IND.EQ.RR) GO TO 500
      C
      COME HERE IF SHAFT

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0030100
0030200
0030300
0030400
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0030700
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0031000
0031100
0031200
0031300
0031400
0031500
0031600
0031700
0031800
0031900
0032000
0032100
0032200
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0032400
0032500
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0032700
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0033000
0033100
0033200
0033300
0033400
0033500
0033600
0033700
0033800
0033900
0034000
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0034400
0034500
0034600
0034700
0034800
0034900
0035000
0035100
0035200
0035300
0035400
0035500
0035600
0035700
0035800
0035900
0036000

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520  WRITE (35,640) IALL(JCX)                0036100
      IMMEC(1,JCXX)=SHAFX                    0036200
      READ (30,1010) IMMEC(2,JCXX)          0036300
      WRITE (35,650)                        0036400
      READ (30,1040) (IMMEC(NH,JCXX),NH=3,6) 0036500
      WRITE (35,660)                        0036600
      READ (30,1040) IMMEC(7,JCXX)          0036700
      KT=18                                0036800
      DO 530 I=1,17                          0036900
530   DESVAL(I,JCXX)=DEFAULT(I,KT)          0037000
540   WRITE (35,670) AA,DESVAL(1,JCXX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2, 0037100
      1,JCXX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,JCXX),RMIN(3,KT),RMAX(3, 0037200
      21),DD,DESVAL(4,JCXX),RMIN(4,KT),RMAX(4,KT),EE,DESVAL(5,JCXX),RMIN( 0037300
      35,KT),RMAX(5,KT),FF,DESVAL(6,JCXX),RMIN(6,KT),RMAX(6,KT),GG,DESVAL 0037400
      4(7,JCXX),RMIN(7,KT),RMAX(7,KT)      0037500
      WRITE (35,960) QQ,RR,TT              0037600
550   CALL REDDEM (IND,DESVAL,JCXX)         0037700
      IF (IND.NE.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 550 0037800
      IF (IND.EQ.QQ) GO TO 570              0037900
      IF (IND.EQ.TT) GO TO 120              0038000
      IF (IND.EQ.RR) GO TO 540              0038100
560   CONTINUE                             0038200
      GO TO 120                             0038300
570   DO 580 I=1,60                         0038400
      IF (IMMEC(1,I).EQ.0) GO TO 580        0038500
      WRITE (6,1180) I,(IMMEC(J,I),J=1,7) 0038600
580   CONTINUE                             0038700
      DO 590 I=1,60                         0038800
      IF (IMMEC(1,I).EQ.0) GO TO 590        0038900
      WRITE (6,1190) I,(DESVAL(J,I),J=1,17) 0039000
590   CONTINUE                             0039100
      WRITE (6,1230)                        0039200
      STOP                                  0039300
C                                           0039400
600   FORMAT ('OMIXER          '," CODE VALUE DESCRIPTION",31X,'LOW VALU 0039500
      1E-HIGH VALUE'/IX,A1,F9.4,' EFF. L/DIAM IF FORCED MIXER ',F20.4,F1 0039600
      20.4/IX,A1,F9.0,' NUMBER OF PASSAGES OR LOBES',F22.0,F10.0) 0039700
610   FORMAT (' COMPONENT ',I3,' IS A NOZZLE, ENTER NOZZLE TYPE'/' 1= CO 0039800
      INVERGENT'/' 2= C-D VARIABLE AREA') 0039900
620   FORMAT ('ENTER THRUST REVERSER TYPE, 0=NONE, 1=FAN, 2=PRIMARY'/' 0040000
      1 ') 0040100
630   FORMAT ('ONNOZZLE        '," CODE VALUE DESCRIPTION",31X,'LOW VALU 0040200
      1E-HIGH VALUE'/IX,A1,F9.4,' LENGTH TO DIAMETER RATIO ',F20.4,F1 0040300
      20.4/IX,A1,F9.4,' BYPASS RATIO FOR MIXED FLOW',F21.4,F10.4) 0040400
640   FORMAT (' COMPONENT ',I3,' IS A SHAFT, ENTER SHAFT NUMBER FROM INN 0040500
      1ER TO OUTER, I.E. 1 2 3 ETC.'/' ') 0040600
650   FORMAT ('ENTER TURBINES CONNECTED TO THIS SHAFT'/' ') 0040700
660   FORMAT ('ENTER COMPONENT NUMBER OF FIRST COMPRESSOR ON THIS SHAFT 0040800
      1'/' ') 0040900
670   FORMAT ('OSHAFT          '," CODE VALUE DESCRIPTION",31X,'LOW VALU 0041000
      1E-HIGH VALUE'/IX,A1,F9.2,' SHAFT ALLOWABLE STRESS ',F20.2,F1 0041100
      20.2/IX,A1,F9.2,' SHAFT MATERIAL DENSITY ',F22.2,F10.2/IX,A1,F 0041200
      39.4,' DIAMETER RATIO (INNER/OUTER)',F21.4,F10.4/IX,A1,F9.0,' C 0041300
      40/COMPONENT NUMBER ON SHAFT FOR INERTIA',F12.0,F10.0)) 0041400
680   FORMAT (' COMPONENT ',I3,' IS A SPLITTER'/' WEIGHT AND LENGTH ARE 0041500
      1GHORED UNLESS FIRST COMPONENT IN THE ENGINE (AFTER INLET)'/' IF I 0041600
      2INNER STREAM IS NOT PRIMARY ENTER A "1" OR ELSE JUST HIT RETURN'/' 0041700
      3 ') 0041800
690   FORMAT (' ENTER MACH NUMBER IN AND HUB TO TIP RATIO IN'/' 0041900
      1 ') 0042000

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700 FORMAT (F6.0,F7.0)
710 FORMAT (' COMPONENT ',I3,' IS A MIXER, ENTER TYPE'/' F- FORCED MIX 0042100
1ER (MECHANICAL - DAISY LOBE TYPE)'/' C- COANNULAR EMERGENCE (NOT F 0042200
2ORCED)') 0042300
720 FORMAT (' ENTER PRIMARY INPUT NODE, 0=INNER, 1=OUTER'/' ') 0042400
730 FORMAT (' TURBINE '/' CODE VALUE DESCRIPTION',31X,'LOW VALU 0042500
1E-HIGH VALUE'/'X,A1,F9.4,' FACE INLET MACH NUMBER ',F20.4,F1 0042600
20.4/'X,A1,F9.5,' LOADING PARAMETER ',F27.5,F10.5/'X,A1,F9.3,' 0042700
3 BLADE SOLIDITY (CORD/SPACING) ',F5.3,F11.3/'X,A1,F9. 0042800
44,' BLADE ASPECT RATIO- 1ST STAGE ',F15.4,F11.4/'X,A1,F9.4,31H 0042900
5 BLADE ASPECT RATIO-LAST STAGE,F19.4,F11.4/'X,A1,F9.4,32H TURBIN 0043000
6E EXIT MACH NUMBER ',F18.4,F11.4/'X,A1,F9.0,31H DISC REFERENC 0043100
7E STRESS ',F19.0,F11.0/'X,A1,F9.0,32H 1=CONST TIP DESN,2=ME 0043200
8AN,3=HUB ',F18.0,F11.0/'X,A1,F9.4,32H MAX. SPEED RATIO RPM/RPMD 0043300
9 ',F18.4,F11.4/'X,A1,F9.4,31H CONTROL RADIUS- 0 IF TRANSFRD,F19.4 0043400
0,F11.4/'X,A1,F9.4,32H BLADE MATERIAL DENSITY 0=T-SET,F18.4,F11.4/ 0043500
1X,A1,F9.4,32H BLADE VOLUME FACTOR ',F18.4,F11.4/'X,A1,F 0043600
09.4,32H BLADE TAPER RATIO ',F18.4,F11.4/'X,A1,F9.4,32H 0043700
0 STATOR BLADE VOLUME FACTOR ',F18.4,F11.4) 0043800
740 FORMAT (' TURBINE '/' CODE VALUE DESCRIPTION',31X,'LOW VALU 0043900
1E-HIGH VALUE'/'X,A1,F9.4,' FACE INLET MACH NUMBER ',F20.4,F1 0044000
20.4/'X,A1,F9.4,' EXIT MACH NUMBER ',F26.4,F10.4) 0044100
750 FORMAT (' COMPONENT ',I3,' IS A HEAT EXCHANGER, OPTIONS ARE'/' F- 0044200
1FIXED OR'/' R- ROTARY') 0044300
760 FORMAT (' FLOW DIRECTION IS'/' P- PARALLEL'/' C- COUNTER FLOW') 0044400
770 FORMAT (' COMPONENT ',I3,' IS A TURBINE, OPTIONS ARE (ENTER CORRE 0044500
1CT LETTER)'/' N- HIGH PRESSURE TURBINE'/' L- LOW PRESSURE TURBINE' 0044600
2) 0044700
780 FORMAT (' A- NO FRAME'/' B- TURBINE HAS EXIT FRAME') 0044800
790 FORMAT (' ENTER COMPONENT NUMBER OF COMPRESSOR THAT SETS TURBINE R 0044900
1PM'/' ') 0045000
800 FORMAT (' ENTER COMPONENT NUMBER THAT LIMITS TURBINE OUTER RADIUS' 0045100
1/' (+ = OUTLET, - = INLET, 0 = FEEDING COMPONENT)') 0045200
810 FORMAT (' ENTER COMPONENT NUMBER THAT LIMITS NOZZLE OUTER RADIUS'/' 0045300
1' (+ = OUTLET, - = INLET, 0 = FEEDING COMPONENT)') 0045400
820 FORMAT (I3) 0045500
830 FORMAT (' YOU MAY ENTER THE NUMBER OF STAGES IF DESIRED, OTHERWISE 0045600
1 ENTER 0'/' ') 0045700
840 FORMAT (' R- RADIAL FLOW TURBINE'/' A- AXIAL FLOW TURBINE') 0045800
850 FORMAT (' INDICATE WHETHER THERE ARE STATORS OR IF THIS IS A CENTR 0045900
1IFUGAL COMPRESSOR'/' A- CALCULATE STATOR WEIGHT'/' B- NO STATOR WE 0046000
2IGHT'/' C- CENTRIFUGAL COMPRESSOR') 0046100
860 FORMAT (' INDICATE WHETHER THERE IS A FRONT FRAME IN THE COMPRESS 0046200
1OR'/' A- NO FRAME'/' B- SNGL BEARING FRAME FOR TFS AND TJS WITHOUT 0046300
2 POWER TAKEOFF (PTO)'/' C- SINGLE BEARING FRAME WITH PTO'/' D- TWO 0046400
3 BEARING FRAME WHICH EXTENDS OUTWARD TOTHE FAN OUTER CASE AND HOLD 0046500
4S TWO BEARINGS WITH PTO') 0046600
870 FORMAT (' INDICATE WHETHER THERE IS A REAR FRAME IN THE COMPRESS 0046700
1OR'/' A- NO FRAME'/' B- SHGL BEARING FRAME FOR TFS AND TJS WITHOUT 0046800
2 POWER TAKEOFF (PTO)'/' C- SINGLE BEARING FRAME WITH PTO'/' D- TWO 0046900
3 BEARING FRAME WHICH EXTENDS OUTWARD TOTHE FAN OUTER CASE AND HOLD 0047000
4S TWO BEARINGS WITH PTO') 0047100
880 FORMAT (' SPLIT FLOW COMPRESSOR - ENTER COMPONENT NUMBER OF OTHER 0047200
1HALF OF COMPRESSOR'/' ') 0047300
890 FORMAT (I2) 0047400
900 FORMAT (' GEAR BOX INDICATOR, 0= NO GEAR BOX, N=SHAFT NUMBER FOR GE 0047500
1AR BOX'/' ') 0047600
910 FORMAT (' YOU MAY SPECIFY THE NUMBER OF STAGES-OVERRIDES MAX PR/ST 0047700
1G., OTHERWISE ENTER 0'/' ') 0047800
920 FORMAT (' HEAT EXCHANGER'/' CODE VALUE DESCRIPTION',31X,'LOW VALU 0047900
0048000

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1E-HIGH VALUE'/IX,A1,F9.0,' NUMBER OF TUBES IF "FIXED" ',F20.0,F1 0048100
20.0/IX,A1,F9.4,' MACH NUMBER - PRIMARY ',F27.4,F10.4/IX,A1,F9.4,' 0048200
3 MACH NUMBER-IN SECONDARY ',F5.4,F11.4/IX,A1,F9. 0048300
44,' BYPASS RATIO IF "ROTARY" ',F15.4,F11.4) 0048400
930 FORMAT ('0FAN/COMPRESSOR'/' CODE VALUE DESCRIPTION',31X,'LOW VALU 0048500
1E-HIGH VALUE'/IX,A1,F9.4,' FACE INLET MACH NUMBER ',F20.4,F1 0048600
20.4/IX,A1,F9.5,' MAX 1ST STAGE PRATIO ',F27.5,F10.5/IX,A1,F9.3,' 0048700
3 COMPRESSOR FACE HUB TO TIP RATIO ',F5.3,F11.3/IX,A1,F9. 0048800
44,' BLADE SOLIDITY (CORD/SPACING) ',F15.4,F11.4/IX,A1,F9.4,31H 0048900
5 BLADE ASPECT RATIO-1ST STAGE ,F19.4,F11.4/IX,A1,F9.4,32H BLADE 0049000
6ASPECT RATIO-LAST STAGE ,F18.4,F11.4/IX,A1,F9.4,31H COMPRESSOR EX 0049100
7IT MACH NUMBER ,F19.4,F11.4/IX,A1,F9.4,32H MAX. COMP. INLET T- 0 0049200
8=CALC'D ,F18.4,F11.4/IX,A1,F9.4,32H MAX. COMP. EXIT T- 0=CALC'D 0049300
9 ,F18.4,F11.4/IX,A1,F9.4,31H MAX. SPEED RATIO RPMMAX/RPMD ,F19.4 0049400
,F11.4/IX,A1,F9.4,32H BLADE MATERIAL DENSITY 0=T-SET,F18.4,F11.4/ 0049500
61X,A1,F9.0,32H 1=CONST.HUB,2=MEAN,3=TIP DESH.,F18.0,F11.0/IX,A1,F 0049600
69.4,32H RPM SCALER TO MATCH KNOWN RPM ,F18.4,F11.4/IX,A1,F9.4,32H 0049700
8 TEMP FOR MATERIAL CHANGE ,F18.4,F11.4/IX,A1,F9.4,31H WEIGH 0049800
6T SCALER 0=AS CALCULATED,F19.4,F11.4/IX,A1,F9.4,32H STATOR BLADE 0049900
6TAPER RATIO ,F18.4,F11.4/IX,A1,F9.4,32H BLADE VOLUME RATIO 0050000
8 ,F18.4,F11.4) 0050100
940 FORMAT ('0CENTRIFUGAL * '/' CODE VALUE DESCRIPTION',31X,'LOW VALU 0050200
1E-HIGH VALUE'/IX,A1,F9.4,' FACE INLET MACH NUMBER ',F22.4,F1 0050300
21.4/IX,A1,F9.5,' MAX 1ST STAGE PRATIO ',F29.5,F10.5/IX,A1,F9.3,' 0050400
3 COMPRESSOR FACE HUB TO TIP RATIO ',F6.3,F11.3/IX,A1,F9 0050500
4.4,' RPM RATIO - RPMMAX/RPMD ',F15.4,F11.4/IX,A1,F9.4,31 0050600
5H COMPRESSOR EXIT MACH NUMBER ,F19.4,F11.4/IX,A1,F9.4,32H GEAR 0050700
6 RATIO OF POWER SHAFT ,F18.4,F11.4/IX,A1,F9.4,31H HORSEPOWER 0 0050800
7F POWER SHAFT ,F19.4,F11.4) 0050900
950 FORMAT (' COMPONENT ',I3,' IS A COMPRESSOR, OPTIONS ARE (ENTER CO 0051000
IRRECT LETTER) '/' A- TYPICAL FAN '/' B- OUTER PORTION OF NON-ROTATIN 0051100
2G SPLITTER FAN '/' C- INNER PORTION OF NON-ROTATING SPLITTER FAN '/' 0051200
3 D- OUTER PORTION OF ROTATING SPLITTER FAN '/' E- INNER PORTION OF 0051300
4ROTATING SPLITTER FAN '/' F- LOW PRESSURE COMPRESSOR '/' G- HIGH PRE 0051400
5SSURE COMPRESSOR') 0051500
960 FORMAT (1H0,'ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POI 0051600
1HT) OR ENTER'/IX,A1,11X,'QUIT PROCESSING ENTIRE ENGINE'/IX,A1,11X, 0051700
2'REVIEW UPDATED VALUES'/IX,A1,11X,'TERMINATE - GO ON TO NEXT COMPO 0051800
3HENT') 0051900
970 FORMAT ('0ENTER VALUES FOR IPLT(PRINTER PLOT), PLOT(GRAPHICS PLOT) 0052000
1, ISII(SI INPUT), ISIO(SI OUTPUT) -- T OR F'/' - - - - -) 0052100
980 FORMAT (4I2) 0052200
990 FORMAT (' &W IPLT=',L2,', PLOT=',L2,', ISII=',L2,', ISIO=',L2,',') 0052300
1000 FORMAT ('0ENTER VALUES FOR INT (2=NO AIRFLOW SCALING,4=WITH AIRFLO 0052400
1W SCALING) AND'/' IOUTCD (0=WT,1, & MAX R,1=ADD COMPONENT WTS,2=A 0052500
2DD STAGE BY STAGE OUTPUT'/' - - -) 0052600
1010 FORMAT (1I,12) 0052700
1020 FORMAT (' INT=',I1,', IOUTCD=',I1,1H,) 0052800
1030 FORMAT ('0ENTER COMPONENT NUMBERS OF ALL COMPONENTS (EXCLUSIVE OF 0052900
1INLETS AND WINJS) THAT CONTRIBUTE TO MAX LENGTH OF ENGINE (RT. ADJ 0053000
2.)'/' - - - - -) 0053100
3 - - - - -) 0053200
1040 FORMAT (12,29I3) 0053300
1050 FORMAT ('0ENTER COMPONENT NUMBERS OF ALL REMAINING COMPONENTS (EXC 0053400
1LUSIVE OF INLETS,WINJS,LOADS,CHTLS,OPTVS, & LIMVS)'/' - - - - -) 0053500
2 - - - - -) 0053600
3 - - - - -) 0053700
1060 FORMAT (' ILENG(1)=' ,30(I2,1H,)) 0053800
1070 FORMAT ('0YOU WILL NOW BE PROMPTED FOR THE GENERIC (NHEP) TYPE FOR 0053900
1 EACH COMPONENT (COMP,THRB, ETC.)') 0054000

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1080 FORMAT ('COMPONENT',I3,' IS A (A4)') 0054100
1090 FORMAT (A4) 0054200
1100 FORMAT ('FOR EACH NNEP COMPONENT YOU WILL BE PROMPTED FOR THE MAT 0054300
1E SUBCOMPONENT') 0054400
1110 FORMAT ('COMPONENT',I3,' IS A DUCT, OPTIONS ARE (ENTER CORRECT L 0054500
1 LETTER)') 0054600
1120 'A- PRIMARY BURNER' 0054700
1130 'B- DUCT BURNER' 0054800
1140 'C- AUGMENTOR' 0054900
1150 'D- DUCT' 0055000
1160 'ENTER LETTER' 0055100
1170 ' ' 0055200
1180 ' ' 0055300
1190 ' ' 0055400
1200 ' ' 0055500
1210 ' ' 0055600
1220 ' ' 0055700
1230 ' ' 0055800
1240 ' ' 0055900
1250 ' ' 0056000
1260 ' ' 0056100
1270 ' ' 0056200
1280 ' ' 0056300
1290 ' ' 0056400
1300 ' ' 0056500
1310 ' ' 0056600
1320 ' ' 0056700
1330 ' ' 0056800
1340 ' ' 0056900
1350 ' ' 0057000
1360 ' ' 0057100
1370 ' ' 0057200
1380 ' ' 0057300
1390 ' ' 0057400
1400 ' ' 0057500
1410 ' ' 0057600
1420 ' ' 0057700
1430 ' ' 0057800
1440 ' ' 0057900
1450 ' ' 0058000
1460 ' ' 0058100
1470 ' ' 0058200
1480 ' ' 0058300
1490 ' ' 0058400
1500 ' ' 0058500
1510 ' ' 0058600
1520 ' ' 0058700
1530 ' ' 0058800
1540 ' ' 0058900
1550 ' ' 0059000
1560 ' ' 0059100
1570 ' ' 0059200
1580 ' ' 0059300
1590 ' ' 0059400
1600 ' ' 0059500
1610 ' ' 0059600
1620 ' ' 0059700
1630 ' ' 0059800
1640 ' ' 0059900
1650 ' ' 0060000

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IF (IX.EQ.GG) DESVAL( 7,JCXX)=VALUE
IF (IX.EQ.HH) DESVAL( 8,JCXX)=VALUE
IF (IX.EQ.II) DESVAL( 9,JCXX)=VALUE
IF (IX.EQ.JJ) DESVAL(10,JCXX)=VALUE
IF (IX.EQ.KK) DESVAL(11,JCXX)=VALUE
IF (IX.EQ.LL) DESVAL(12,JCXX)=VALUE
IF (IX.EQ.MM) DESVAL(13,JCXX)=VALUE
IF (IX.EQ.NN) DESVAL(14,JCXX)=VALUE
IF (IX.EQ.OO) DESVAL(15,JCXX)=VALUE
IF (IX.EQ.PP) DESVAL(16,JCXX)=VALUE
IF (IX.EQ.SS) DESVAL(17,JCXX)=VALUE
RETURN
END
```

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0060100
0060200
0060300
0060400
0060500
0060600
0060700
0060800
0060900
0061000
0061100
0061200
0061300
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#### REFERENCES

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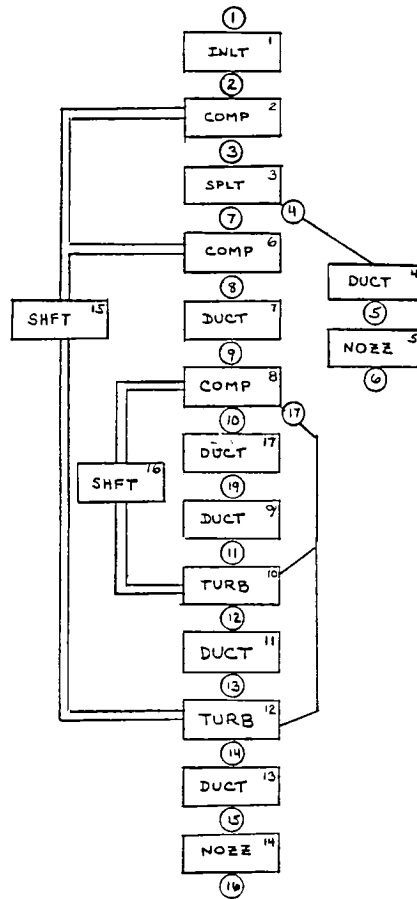


Figure 1. - Thermodynamic schematic for NNEP.

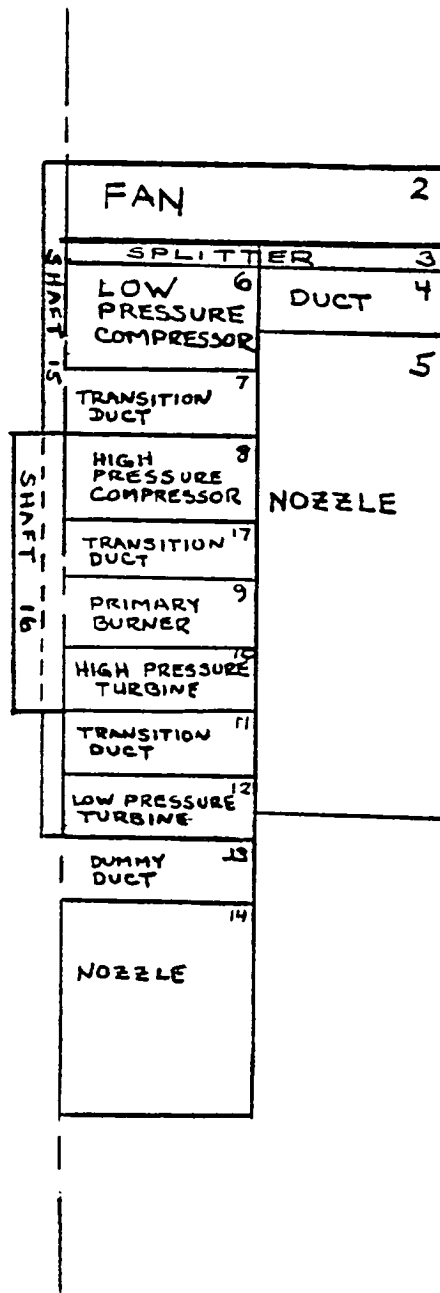


Figure 2. - "Actual" mechanical schematic for WATE by subcomponent type.

COMP#	TYPE	NSTAGE	WT
1	INLT	0	0.
2	FAN	1	1455.
3	SPLY	0	0.
4	LPC	1	314.
7	DUCT	0	12.
8	HPC	14	582.
17	DUCT	0	22.
9	PBUR	0	329.
10	HPT	2	159.
11	DUCT	0	8.
12	LPT	3	1189.
13	DUCT	0	0.
14	NOZ	0	156.
4	DUCT	0	12.
5	NOZ	0	243.

TOTAL ENGINE WT= 4800.

SAMPLE CASE FOR PREPWATE

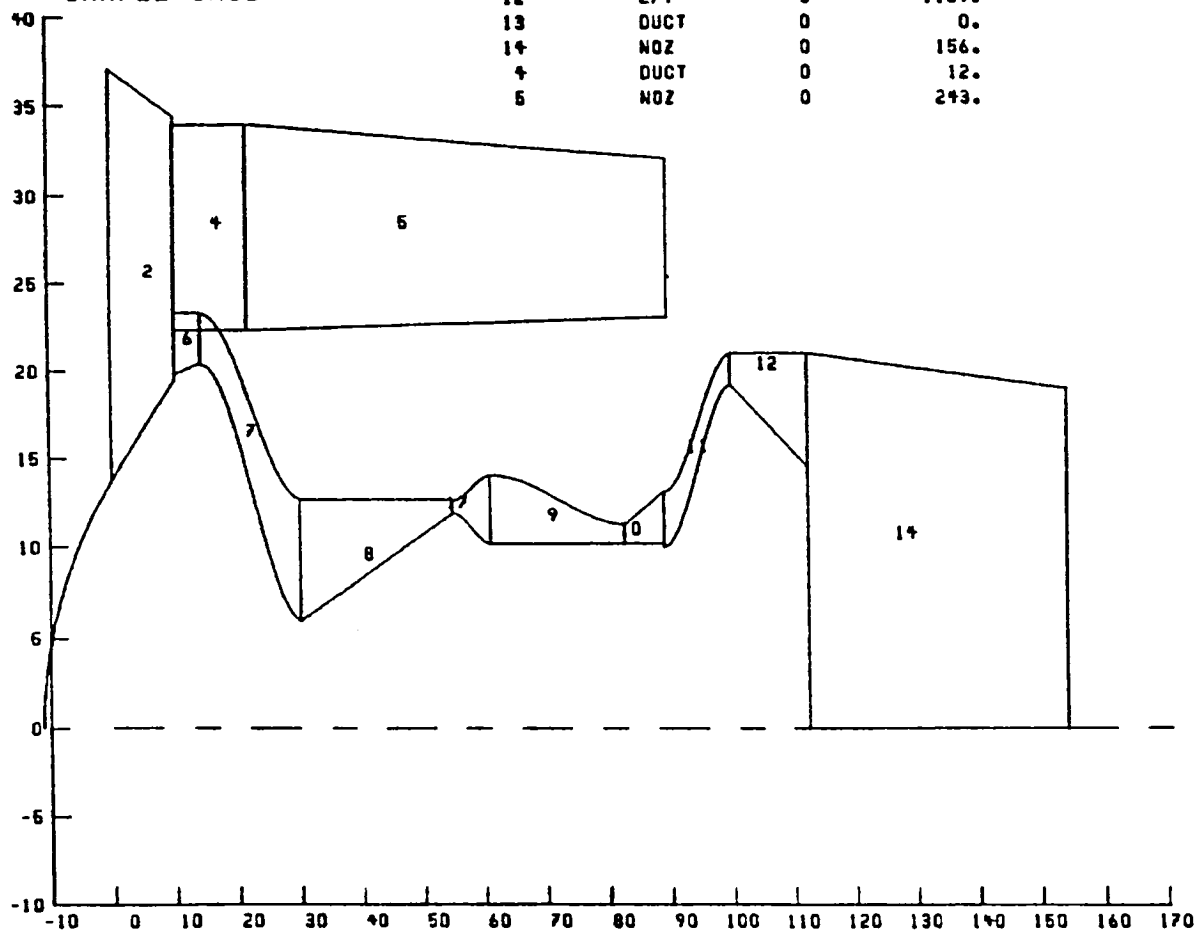


Figure 3. - Graphics plot of engine flowpath.

1. Report No. NASA TM-83545		2. Government Accession No.		3. Recipient's Catalog No.	
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16. Abstract <p>The Weight Analysis of Turbine Engines (WATE) computer code was developed by Boeing under contract to NASA Lewis (ref. 1). It was designed to function as an adjunct to the Navy/NASA Engine Program (NNEP) (ref. 2). NNEP calculates the design and off-design thrust and sfc performance of User defined engine cycles. The thermodynamic parameters throughout the engine as generated by NNEP are then combined with input parameters defining the component characteristics in WATE to calculate the bare engine weight of this User defined engine. Preprocessor programs for NNEP were previously developed (ref. 3) to simplify the task of creating input datasets. This report describes a similar preprocessor for the WATE code.</p>					
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